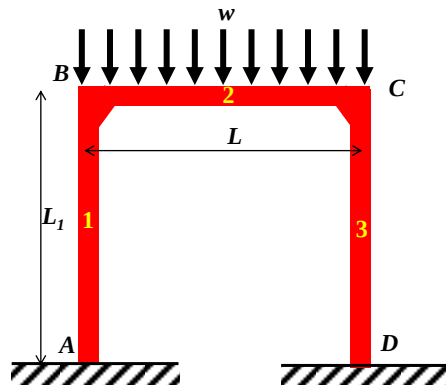


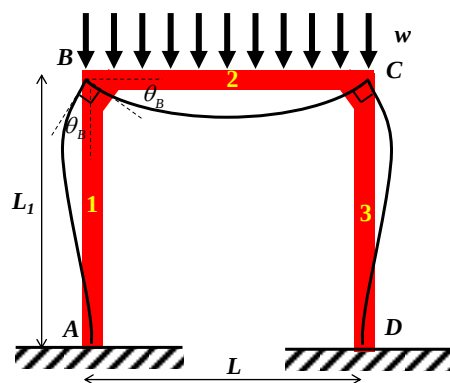
Indeterminate Frame:
Full Moment connections



Unknowns: Moment and forces transmitted through each joint

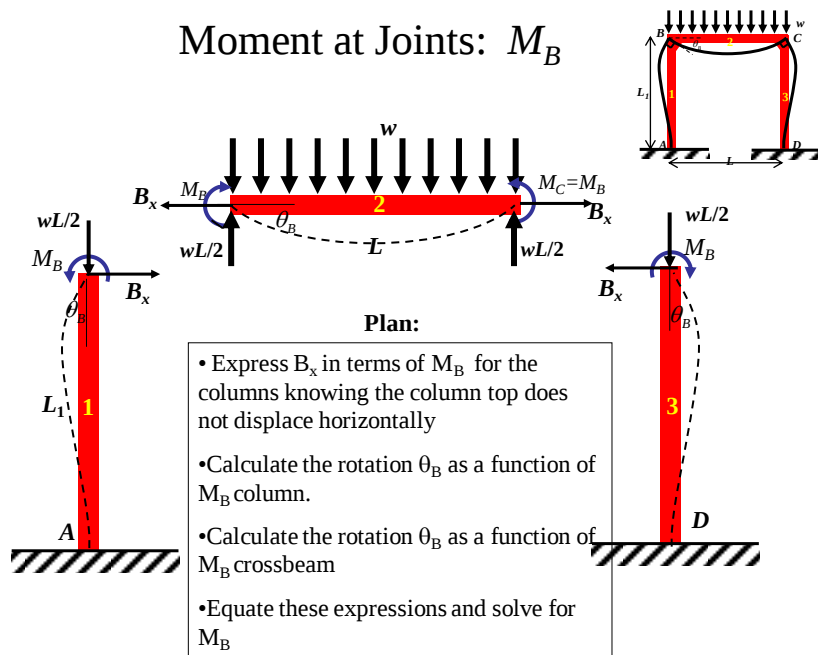
Equations: 3 per member

Full Moment connections:
Joint angles preserved

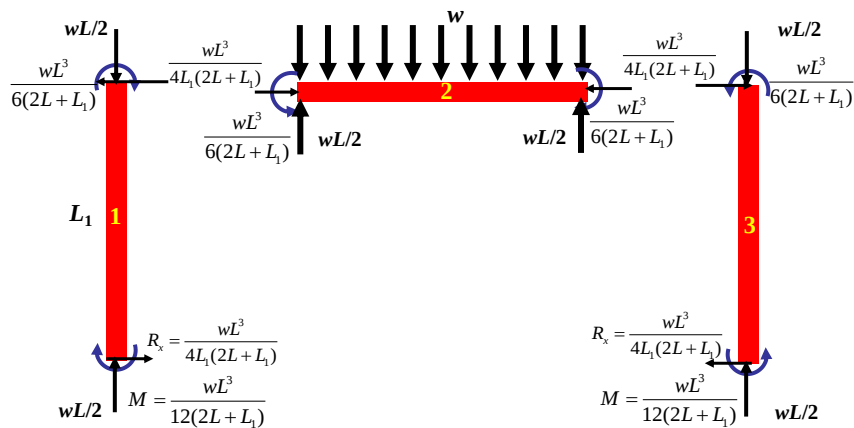


Neglect Axial Deformation: $u^B = u^C = 0$

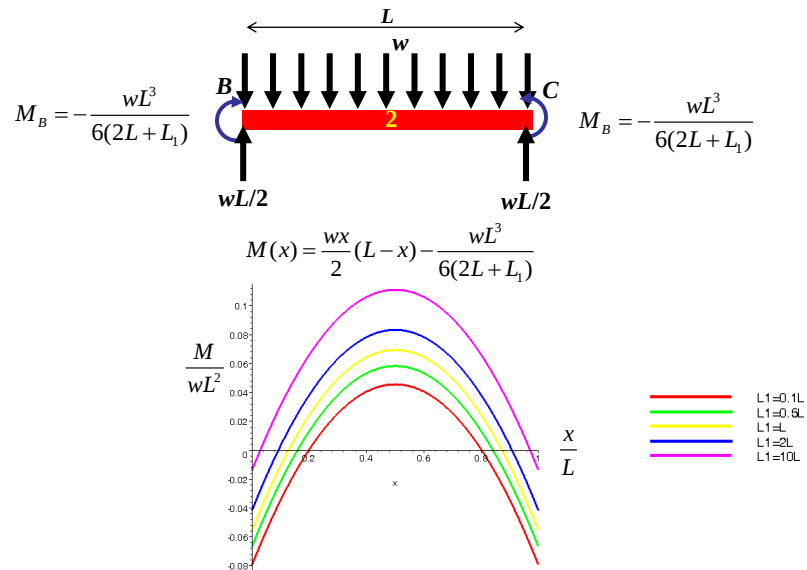
Moment at Joints: M_B



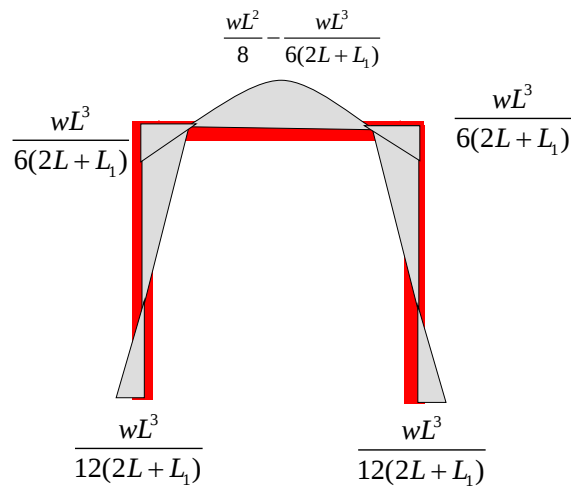
Finally



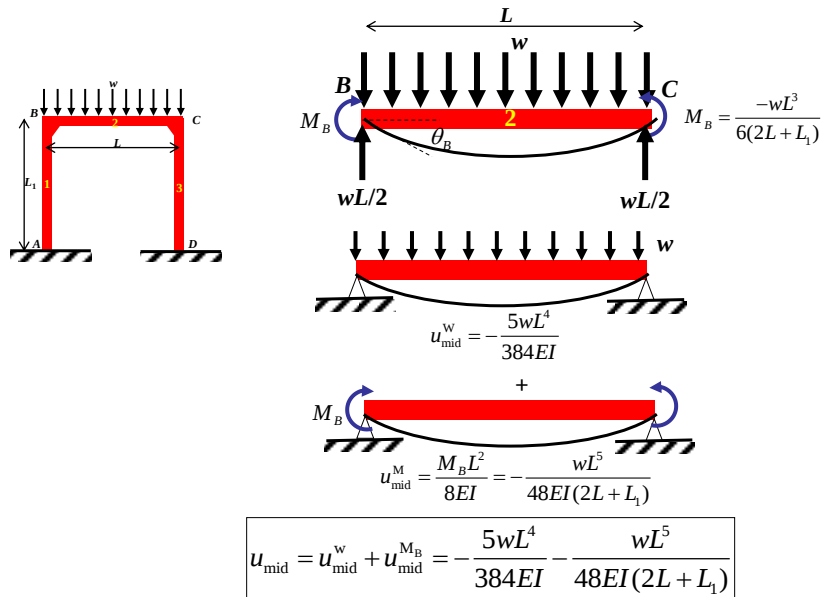
Moments in the cross beam



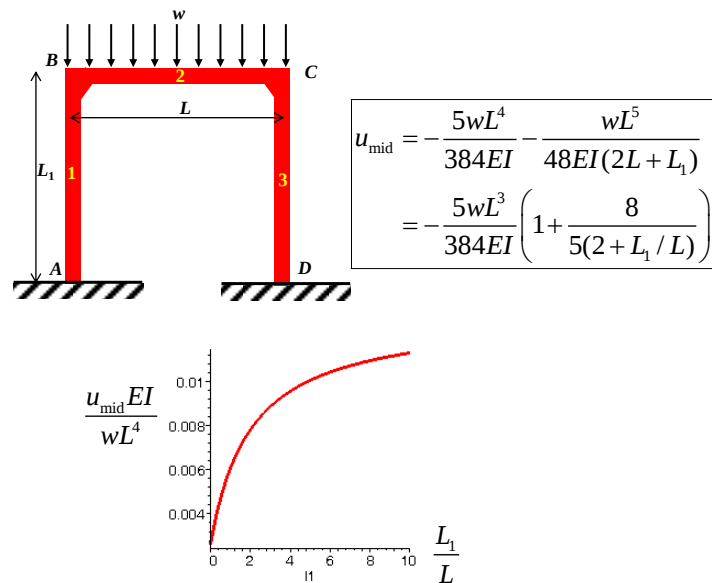
Moment Diagram



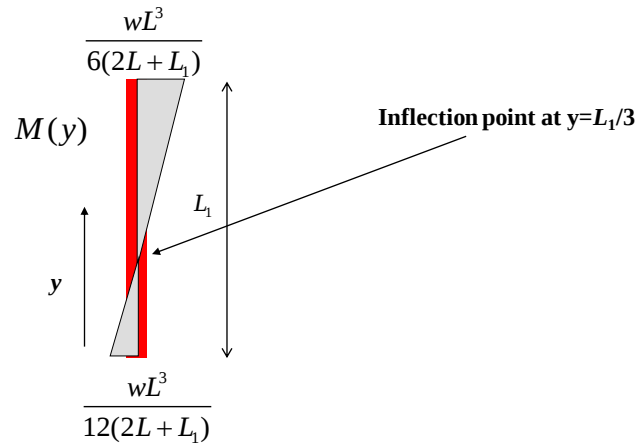
Deflection at Midspan



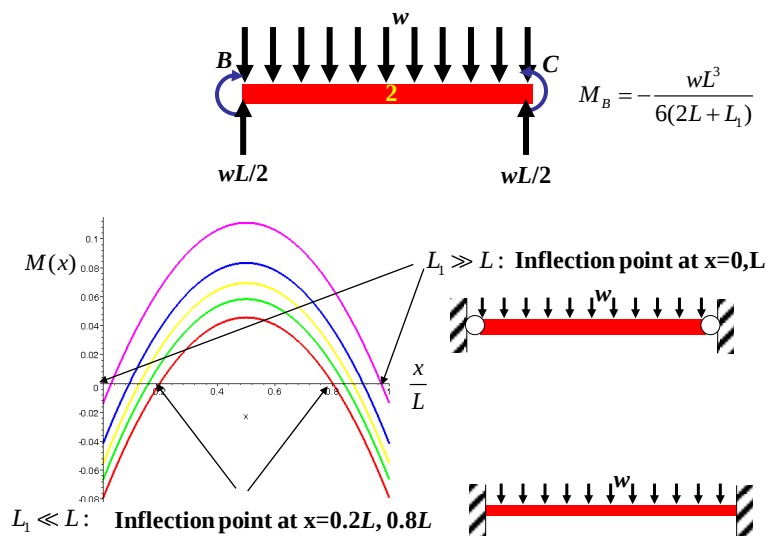
Deflection at Midspan



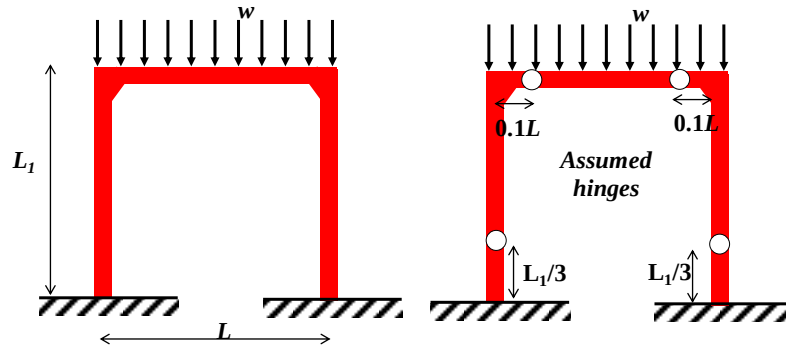
Inflection Points: Columns



Inflection Points: Cross Beam



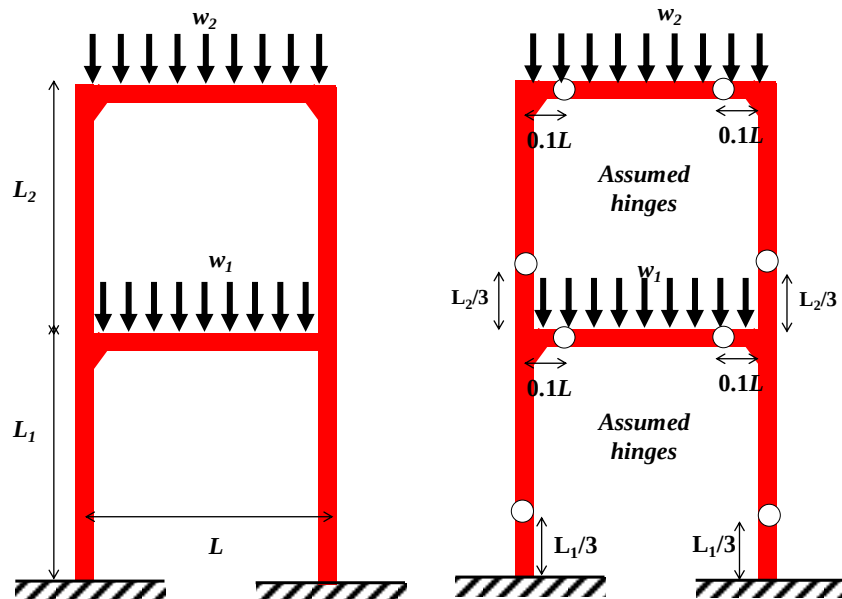
Approximate Analysis:
Place inflection points at $x=0.1L$, $0.9L$, $y=L_1/3$



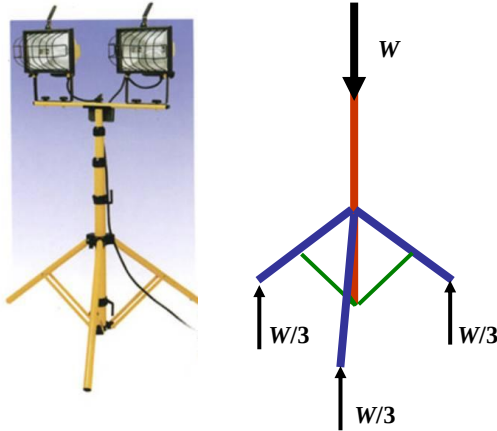
May be used to approximate situations with:

- $EI_{\text{columns}} \neq EI_{\text{crossbeam}}$
- No full moment connections at the corners
- Multi-unit structures
- Vertical loads relatively close to those shown above (no horizontal loads)

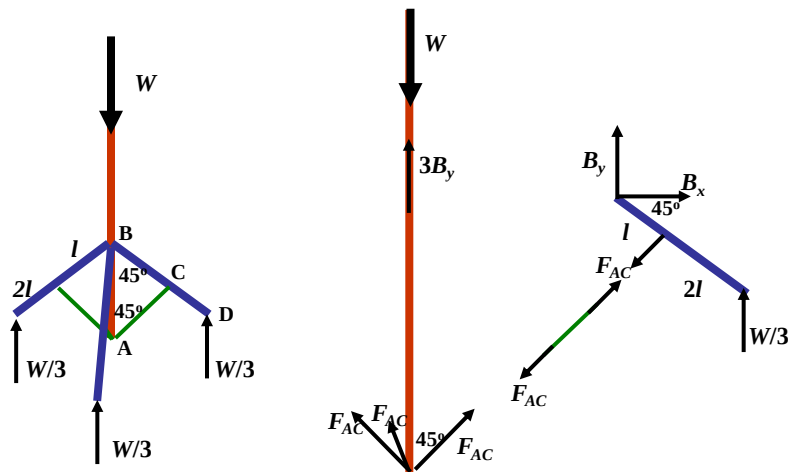
Multistory



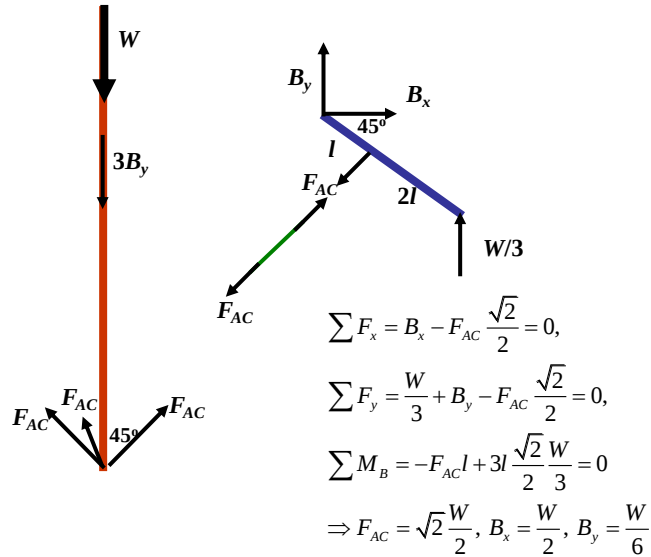
Example: Folding Stand



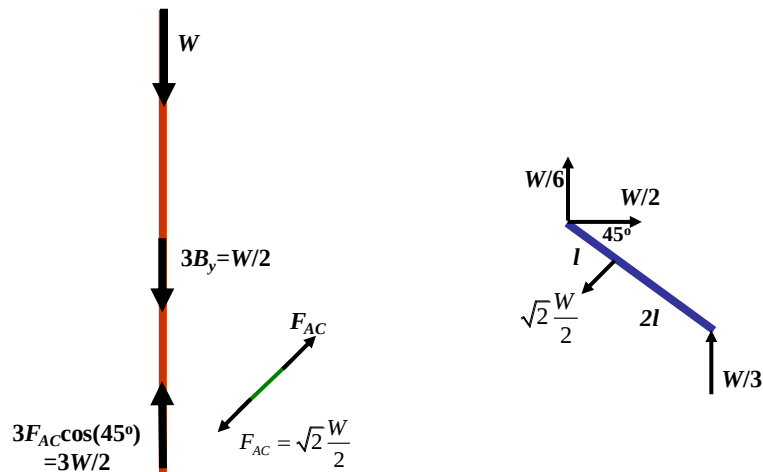
Some Disassembly Required!



Example



Internal forces:



Deflections?

Find the deflection of the poles due to bending of the legs.

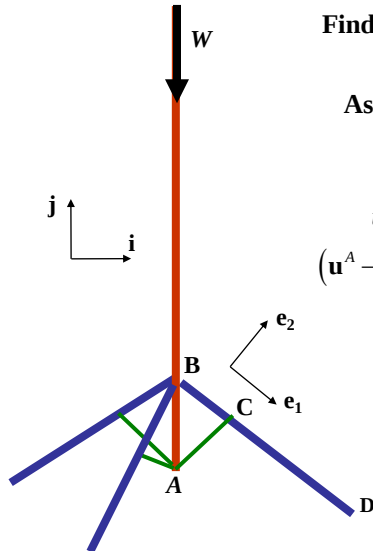
Assume no axial extension in any member.

$$u_x^A = u_x^B = 0 \text{ (symmetry)}$$

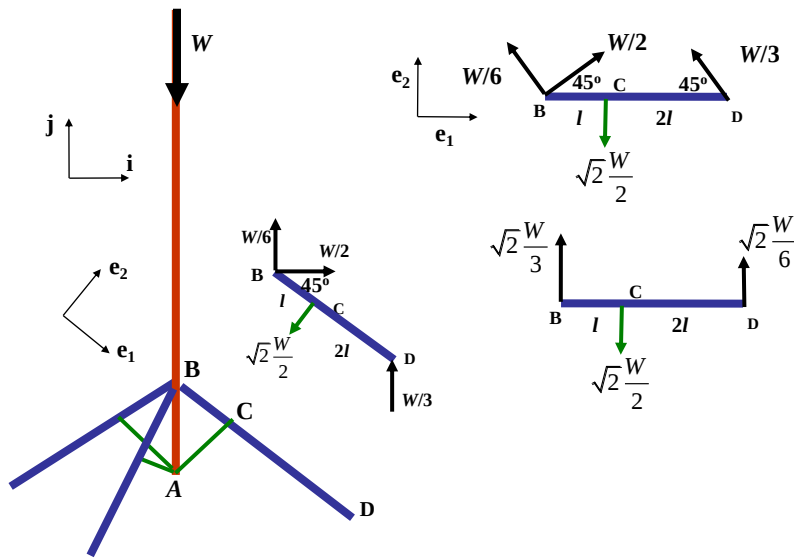
$$u_y^B = u_y^A \text{ (no axial deformation of the pole)}$$

$$(\mathbf{u}^A - \mathbf{u}^C) \cdot \mathbf{n}^{CA} = 0 \text{ (no axial deformation of the strut)}$$

$$u_y^D = 0 \text{ (Support at the base)}$$



A Leg



A Leg

$M(x) = \sqrt{2} \frac{W}{3} x, (0 < x < l),$
 $M(x) = \sqrt{2} \frac{W}{6} (l - x) (l < x < 3l)$

$\frac{d^2 v}{dx^2} = \frac{M(x)}{EI}$

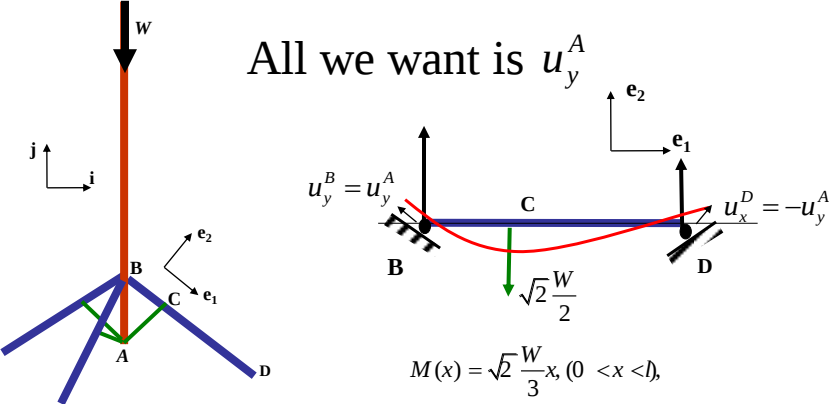
What are the Boundary Conditions on v ?

These...

$u_x^A = u_x^B = 0$ (symmetry)
 $u_y^B = u_y^A$ (no axial deformation of the pole)
 $(\mathbf{u}^A - \mathbf{u}^C) \cdot \mathbf{n}^{CA} = 0$ (no axial deformation of the strut)
 $u_y^D = 0$ (Support at the base)

$(\mathbf{u}^A - \mathbf{u}^C) \cdot \mathbf{e}_2 = 0 \Rightarrow u_y^A \frac{\sqrt{2}}{2} = u_2^C = v(l)$
 $u_y^B = -u_x^D$

All we want is u_y^A

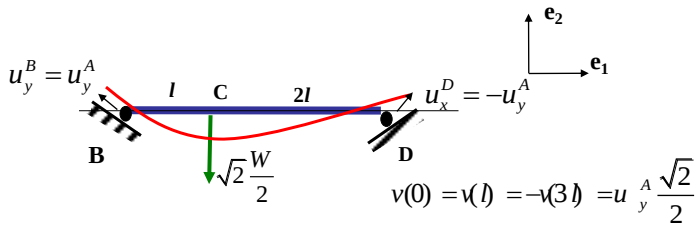


$$M(x) = \sqrt{2} \frac{W}{3} x \quad (0 < x < l)$$

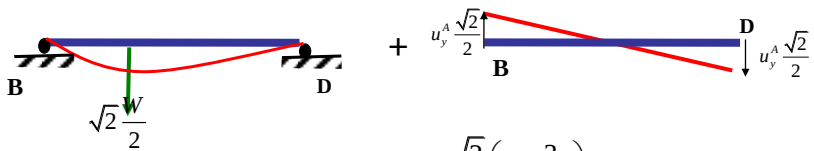
$$M(x) = \sqrt{2} \frac{W}{6} (l - x) \quad (l < x < 3l)$$

$$\frac{d^2 v}{dx^2} = \frac{M(x)}{EI} \quad v(0) = v(l) = -v(3l) = u_y^A \frac{\sqrt{2}}{2}$$

How?

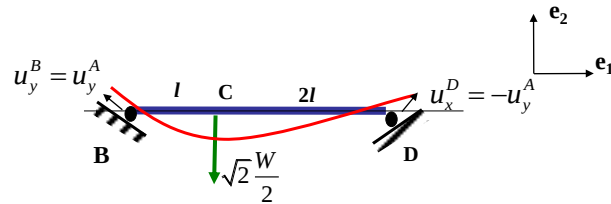


Simply supported beam deflection + *Rigid Rotation about the center*



$$v(x) = v_{ss}(x) - u_y^A \frac{\sqrt{2}}{3l} \left(x - \frac{3}{2}l \right)$$

Why?



$$v_c = v(l) = v_{ss}(l) - u_y^A \frac{\sqrt{2}}{3l} \left(l - \frac{3}{2}l \right) = v_{ss}(l) + u_y^A \frac{\sqrt{2}}{6} = u_y^A \frac{\sqrt{2}}{2}$$

$$v_{ss}(l) = u_y^A \frac{\sqrt{2}}{3} = -\frac{4}{9} \frac{Pl^3}{EI} \Rightarrow u_y^A = -2\sqrt{2} \frac{Pl^3}{3EI}$$

Vertical deflection is $u_y^A = -2\sqrt{2} \frac{Pl^3}{3EI}$

