## School of Engineering Brown University

## **EN1750: Advanced Mechanics of Solids**

## Homework 4: Internal forces and equations of motion.

Due Friday Oct 5, 2018

1. For the Cauchy stress tensor with components

$$\mathbf{\sigma} = \begin{bmatrix} 50 & 250 & 0 \\ 250 & 130 & -10 \\ 0 & -10 & 320 \end{bmatrix}$$

compute

- 1. The traction vector acting on an internal material plane with normal  $\mathbf{n} = \frac{1}{\sqrt{2}} \mathbf{e}_1 \frac{1}{\sqrt{2}} \mathbf{e}_2$
- 2. The principal stresses
- 3. The hydrostatic stress
- 4. The deviatoric stress tensor
- 5. The Von-Mises equivalent stress (find the answer using the answers to both (2) and (4))

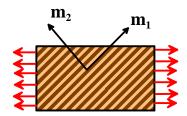
You can use MATLAB for this – if you do, just paste a screenshot of your matlab script and its solutions into the pdf or word document you use to submit your solutions.

This is just a matter of using the definitions. The numbers are best crunched using MATLAB. A solution is shown below

traction = 
$$(-100 \sqrt{2} \ 60 \sqrt{2} \ 5 \sqrt{2})$$
  
ps =  $3\times1$   
-163.2669  
317.8057  
345.4612  
hydrostatic =  $166.6667$   
sdev =  $3\times3$   
-116.6667 250.0000 0  
250.0000 -36.6667 -10.0000  
0 -10.0000 153.3333  
smises =  $495.4796$   
smises =  $495.4796$ 

[5 POINTS]

**2.** The 'Tsai-Hill' criterion is used to predict the critical loads that cause failure in brittle laminated fiber-reinforced composites and wood. A specimen of laminated composite subjected to in-plane loading is sketched in the figure. The Tsai-Hill criterion assumes that a plane stress state exists in the solid. Let  $\sigma_{11}$ ,  $\sigma_{22}$ ,  $\sigma_{12}$  denote the nonzero components of stress, with basis vectors  $\mathbf{m}_1$  and  $\mathbf{m}_2$  oriented parallel and perpendicular to the fibers in the sheet, as shown. The Tsai-Hill failure criterion is



$$\left(\frac{\sigma_{11}}{\sigma_{TS1}}\right)^2 + \left(\frac{\sigma_{22}}{\sigma_{TS2}}\right)^2 - \frac{\sigma_{11}\sigma_{22}}{\sigma_{TS1}^2} + \frac{\sigma_{12}^2}{\sigma_{SS}^2} = 1$$

at failure, where  $\sigma_{TS1}$ ,  $\sigma_{TS2}$  and  $\sigma_{SS}$  are material properties. (The material fails when the quantity on the left hand side of the equation is equal to 1). This report gives the following data for the properties of a graphite-epoxy CFRP

Property	Value (MPa)
$\sigma_{\scriptscriptstyle TS1}$	2206
$\sigma_{\scriptscriptstyle TS2}$	56.5
$\sigma_{\scriptscriptstyle SS}$	110.3

Suppose that a specimen is loaded in uniaxial tension with the tensile axis at 45 degrees to the fiber direction, as shown in the figure. Calculate the maximum stress that the material can withstand (you will need to use the basis change formulas for a tensor).

The general formula relating tensor components in two bases is

$$\begin{bmatrix} S_{11}^{(\mathbf{m})} & S_{12}^{(\mathbf{m})} & S_{13}^{(\mathbf{m})} \\ S_{21}^{(\mathbf{m})} & S_{22}^{(\mathbf{m})} & S_{23}^{(\mathbf{m})} \\ S_{31}^{(\mathbf{m})} & S_{32}^{(\mathbf{m})} & S_{33}^{(\mathbf{m})} \end{bmatrix} = \begin{bmatrix} \mathbf{m}_1 \cdot \mathbf{e}_1 & \mathbf{m}_1 \cdot \mathbf{e}_2 & \mathbf{m}_1 \cdot \mathbf{e}_3 \\ \mathbf{m}_2 \cdot \mathbf{e}_1 & \mathbf{m}_2 \cdot \mathbf{e}_2 & \mathbf{m}_2 \cdot \mathbf{e}_3 \\ \mathbf{m}_3 \cdot \mathbf{e}_1 & \mathbf{m}_3 \cdot \mathbf{e}_2 & \mathbf{m}_3 \cdot \mathbf{e}_3 \end{bmatrix} \begin{bmatrix} S_{11}^{(\mathbf{e})} & S_{12}^{(\mathbf{e})} & S_{13}^{(\mathbf{e})} \\ S_{21}^{(\mathbf{e})} & S_{22}^{(\mathbf{e})} & S_{23}^{(\mathbf{e})} \\ S_{31}^{(\mathbf{e})} & S_{32}^{(\mathbf{e})} & S_{33}^{(\mathbf{e})} \end{bmatrix} \begin{bmatrix} \mathbf{m}_1 \cdot \mathbf{e}_1 & \mathbf{m}_2 \cdot \mathbf{e}_1 & \mathbf{m}_3 \cdot \mathbf{e}_1 \\ \mathbf{m}_1 \cdot \mathbf{e}_2 & \mathbf{m}_2 \cdot \mathbf{e}_2 & \mathbf{m}_3 \cdot \mathbf{e}_2 \\ \mathbf{m}_3 \cdot \mathbf{e}_1 & \mathbf{m}_3 \cdot \mathbf{e}_3 \end{bmatrix}$$

Here

$$\frac{\sigma}{2} \begin{bmatrix} 1 & -1 & 0 \\ -1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} 1/\sqrt{2} & 1/\sqrt{2} & 0 \\ -1/\sqrt{2} & 1/\sqrt{2} & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} \sigma & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} 1/\sqrt{2} & -1/\sqrt{2} & 0 \\ 1/\sqrt{2} & 1/\sqrt{2} & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

Hence  $\sigma_{11} = \sigma_{22} = \sigma / 2$   $\sigma_{12} = -\sigma / 2$ 

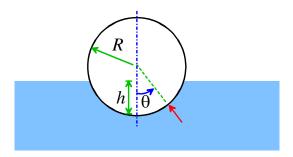
$$\left(\frac{\sigma/2}{\sigma_{TS1}}\right)^{2} + \left(\frac{\sigma/2}{\sigma_{TS2}}\right)^{2} - \frac{\sigma^{2}/4}{\sigma_{TS1}^{2}} + \frac{\sigma^{2}/4}{\sigma_{SS}^{2}} = 1$$

$$\Rightarrow \frac{\sigma^{2}}{4} \left(\frac{1}{\sigma_{TS2}^{2}} + \frac{1}{\sigma_{SS}^{2}}\right) = 1$$

$$\Rightarrow \sigma = 100.6MPa$$

[5 POINTS]

**3.** A cylinder of radius *R* is partially immersed in a static fluid.



(3.1) Recall that the pressure at a depth d in a fluid has magnitude rgd. Write down an expression for the horizontal and vertical components of traction acting on the surface of the cylinder in terms of q.

The pressure is 
$$p = \rho g \left[ h - R (1 - \cos \theta) \right]$$

The traction acts radially so

$$t_1 = -p\sin\theta = -\rho g(h - R(1 - \cos\theta))\sin\theta$$

$$t_2 = p\cos\theta = \rho g (h - R(1 - \cos\theta))\cos\theta$$

[2 POINTS]

(3.2) Hence compute the resultant force (per unit out of plane distance) exerted by the fluid on the cylinder, in terms of  $\rho$ , g, h, R.

The total force will be

$$F_1 = \int_{-\phi}^{\phi} t_1 R d\theta$$
  $F_2 = \int_{-\phi}^{\phi} t_2 R d\theta$   $\phi = \cos^{-1}(1 - h/R)$ 

Evaluating the integrals gives

$$F_1 = 0$$
 (obvious)

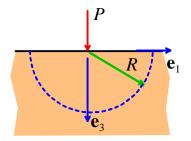
$$F_2 = \rho g \left[ R^2 \cos^{-1} \left( 1 - \frac{h}{R} \right) - (R - h) \sqrt{R^2 - (R - h)^2} \right]$$

You can check the answer by setting h=R – in this case the volume of fluid displaced is obviously  $\pi R^2/2$ 

[3 POINTS]

**4.** The stresses in a large incompressible elastic solid with a flat surface, which has a point force magnitude P acting vertically downwards at the origin is given by

$$\sigma_{ij} = \frac{-3Px_i x_j x_3}{2\pi R^5}$$



4.1 Show that the stress state satisfies the equilibrium equation

$$\frac{\partial \sigma_{ij}}{\partial x_i} = 0$$

Recall that

$$\frac{\partial R}{\partial x_i} = \frac{x_i}{R} \qquad \frac{\partial x_i}{\partial x_j} = \delta_{ij}$$

Therefore

$$\frac{\sigma_{ij}}{\partial x_i} = \frac{-3P}{2\pi R^5} \left( \delta_{ii} x_j x_3 + x_i \delta_{ij} x_3 + x_i x_j \delta_{i3} - \frac{x_i x_j x_3 x_i}{R^2} \right) = 0$$

Where we noted that  $\delta_{ii} = 3$   $x_i x_i = R^2$ 

[3 POINTS]

4.2 Find a formula for the traction acting on a hemispherical surface with radius R inside the solid (note that  $x_i / R$  is a unit vector normal to the surface).

$$t_j = n_i \sigma_{ij} = \frac{x_i}{R} \frac{-3Px_i x_j x_3}{2\pi R^5} = \frac{-3Px_j x_3}{2\pi R^4}$$

[2 POINTS]

**5. Importing CAD data into ABAQUS**. Work through the tutorial on this topic. Then, use Solidworks to create a simple part, import it into ABAQUS (try both a step or iges import and a .inp file import. Its best to avoid the STL file import), and run a simulation. Keep everything simple, or you will run into problems. Try to make your part differ from those submitted by your classmates!

Please upload your solidworks model and cae file to canvas as a solution to this problem.

[20 POINTS]