

Failure under monotonic loading

1. Brittle failure – little permanent deformation prior to failure; faceted failure surface
2. Ductile failure – extensive permanent deformation prior to failure; dimpled fracture surface

Brittle failure criteria

$\{\sigma_1, \sigma_2, \sigma_3\}$ Principal stresses $\sigma_1 > \sigma_2 > \sigma_3$

$$\tau = (\sigma_1 - \sigma_3) / 2 \quad \sigma_m = (\sigma_1 + \sigma_3) / 2$$

Isotropic failure criteria

$\sigma_1 > 0$ Failure by fracture when $\sigma_1 = \sigma_f$

$\sigma_1 < 0$ Failure by crushing (eg Mohr Coulomb criterion)

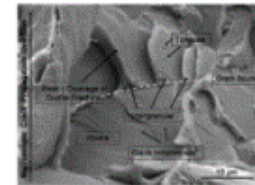
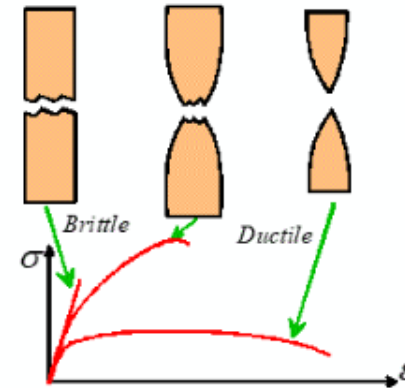
Anisotropic failure criteria (eg Tsai-Hill for composite laminates)

Basis parallel/perpendicular to fibers $\{e_1, e_2, e_3\}$

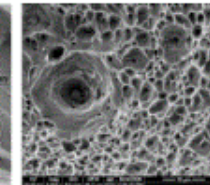
Plane stress state $\{\sigma_{11}, \sigma_{22}, \sigma_{12}\}$

Failure criterion

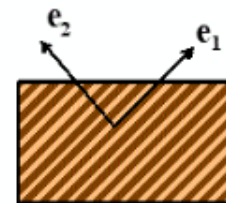
$$\left(\frac{\sigma_{11}}{\sigma_{TS1}}\right)^2 + \left(\frac{\sigma_{22}}{\sigma_{TS2}}\right)^2 - \frac{\sigma_{11}\sigma_{22}}{\sigma_{TS1}\sigma_{TS2}} + \frac{\sigma_{12}}{\sigma_{SS}} = 1$$



Cleavage fracture (brittle)



Dimpled rupture (ductile)



Probabilistic Brittle Failure Criterion (Weibull statistics)

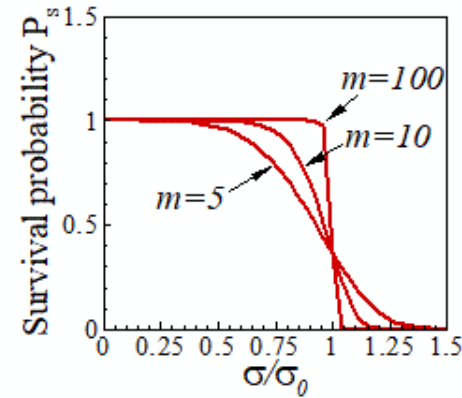
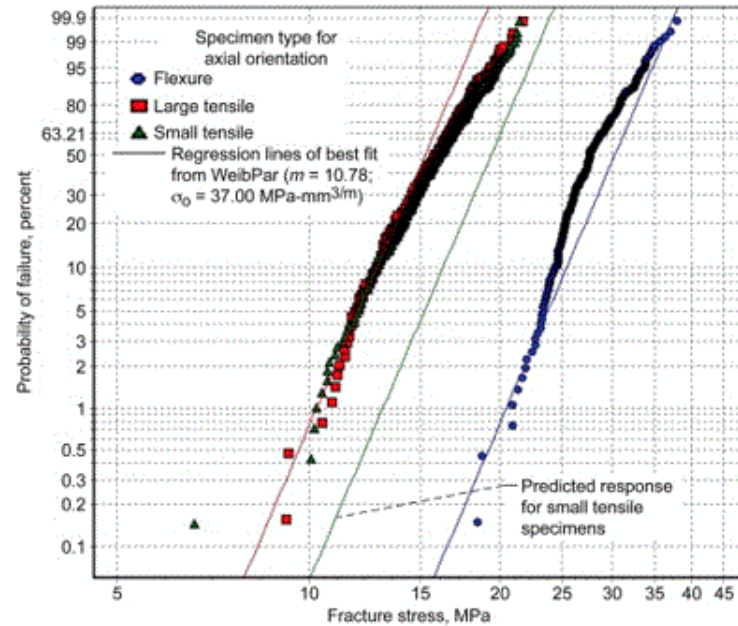
Experimental Observations

- (1) Large scatter in failure stress
- (2) Failure stress depends on specimen volume & geometry

Approach to predicting failure: fit survival probability with "Weibull distribution"

$$P_s = \exp \left\{ - \frac{V}{V_0} \left(\frac{\sigma}{\sigma_0} \right)^m \right\}$$

(σ_0, m) material props
 V_0 - specimen volume



Fracture statistics in graphite <https://doi.org/10.1016/j.carbon.2013.02.054>

Measuring σ_0, m

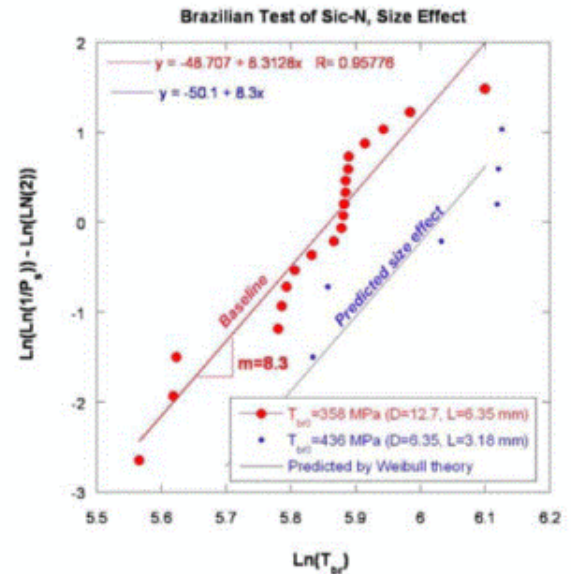
Do large number of tests; measure P_s as function of σ for specimens with volume $V = V_0$

To find σ_0, m note $\log P_s = -\frac{V}{V_0} \left(\frac{\sigma}{\sigma_0}\right)^m = -\left(\frac{\sigma}{\sigma_0}\right)^m$

$$\Rightarrow \log\left(\log\left(\frac{1}{P_s}\right)\right) = m \log \sigma - m \log \sigma_0$$

Slope gives m

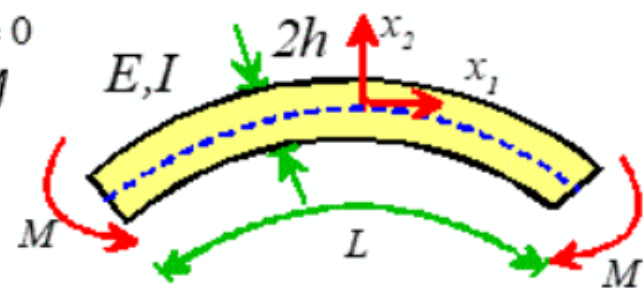
Intercept gives $m \log \sigma_0$



Example: A brittle beam with Young's modulus E , inertia $I_{22} = I_{11} = I$, $I_{12} = 0$ and length L and rectangular cross-section $2h \times b$ is bent by a moment M

The beam has Weibull parameters V_0, σ_0, m

Find a formula for the probability for the beam to survive



Use formula: Extend to multiaxial loading with nonuniform stress

Survival probability

$$\log(P_s) = - \frac{1}{V_0 \sigma_0^m} \int_V (\langle \sigma_1 \rangle^m + \langle \sigma_2 \rangle^m + \langle \sigma_3 \rangle^m) dV$$

Where $(\sigma_1, \sigma_2, \sigma_3)$ are principal stresses

$$\langle x \rangle = \begin{cases} x & x \geq 0 \\ 0 & x \leq 0 \end{cases}$$

Apply to beam $\sigma_{11} = \frac{M x_2}{I}$ all other $\sigma_{ij} = 0$

$$\begin{aligned} \text{Hence } \log(P_s) &= -\frac{1}{\bar{V}_0 \sigma_0^m} \int_0^b \int_0^L \int_{-h}^h \left\langle \frac{M x_2}{I} \right\rangle^m dx_2 dx_1 dx_3 \\ &= -\frac{\underbrace{b L h}_{\bar{V}/2}}{\bar{V}_0 \sigma_0^m} \frac{1}{(m+1)} \left(\frac{M h}{I} \right)^m \quad \text{note } \frac{M h}{I} = \sigma_{\max} \end{aligned}$$

$$P_s = \exp \left\{ -\frac{\bar{V}}{2 \bar{V}_0 (m+1)} \left(\frac{\sigma_{\max}}{\sigma_0} \right)^m \right\}$$

Survival probability in bending > survival probability in tension

Failure criteria for ductile materials

Failure occurs by nucleation & growth of voids; driven by plastic strain
 Voids grow faster in regions of high hydrostatic tension

(1) Simple strain based failure criterion

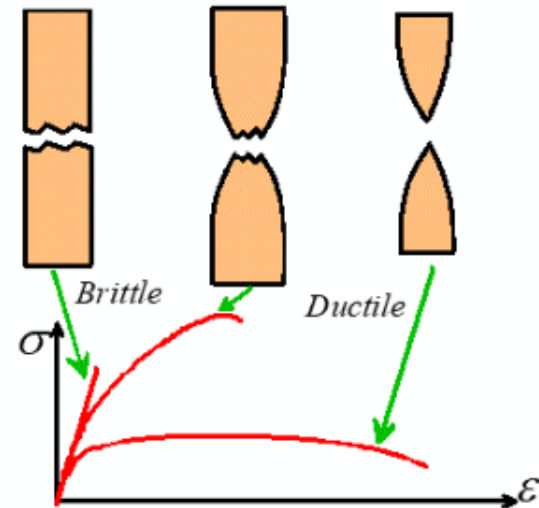
Let $\epsilon_e =$ Von Mises plastic strain

$\epsilon_e < \epsilon_f$ safe
 $\epsilon_e = \epsilon_f$ fail

ϵ_f material prop

Use torsion test to measure ϵ_f

[Not very accurate]



(2) Johnson - Cook failure model

- accounts for temperature, strain rate, hydrostatic stress

Let $\bar{\sigma}_e$ - Von-Mises stress

$$\bar{\sigma}_h = \bar{\sigma}_{kk}/3 \quad : \text{hydrostatic stress}$$

$\dot{\epsilon}_e$ - Von Mises plastic strain rate

Define "Damage parameter" D

$$D = \int_0^t \frac{\dot{\epsilon}_e dt}{\left[d_1 + d_2 \exp\left(\frac{d_3 \bar{\sigma}_h}{\bar{\sigma}_e}\right) \right] \left[1 + d_4 \frac{\dot{\epsilon}_e}{\dot{\epsilon}_0} \right]}$$

$(\dot{\epsilon}_0, d_1, d_2, d_3, d_4)$ material properties

Lots of experiments needed to determine these

Failure criterion

$D < 1$ everywhere safe

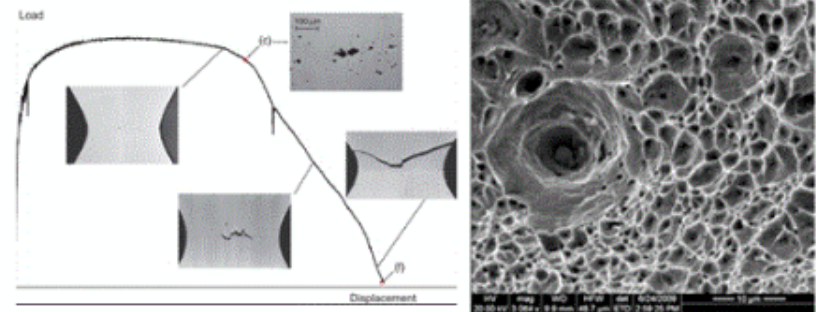
$D = 1$ fail

ABAQUS can delete elements when $D = 1$
to simulate fracture

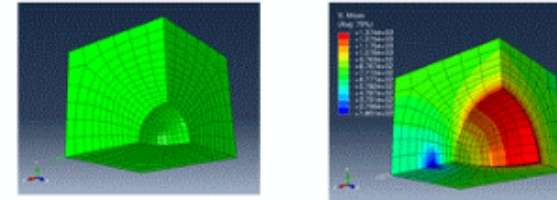
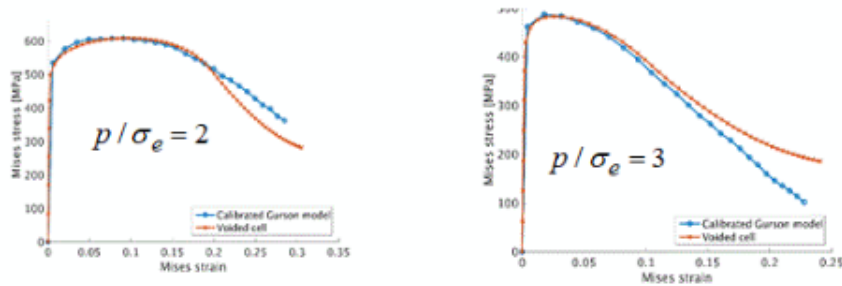
Porous plasticity 'Gurson' model in ABAQUS

Failure mechanism in ductile metals

1. Specimen necks
2. Hydrostatic (tensile) stress increases in neck
3. Voids nucleate at 2nd phase particles
4. Voids grow
5. Voids coalesce to form crack



Simulations of void growth



Gurson (porous plasticity model)

Plastic strain rate magnitude $\dot{\epsilon}_e = g(\sigma_e, p, \sigma_0, f^*) = \dot{\epsilon}_0 \left[\left(\frac{\sigma_e}{Y(\epsilon_e)} \right)^2 + 2q_1 f^* \cosh \left(q_2 \frac{3p}{2Y(\epsilon_e)} \right) - (q_1 f^*)^2 - 1 \right]^{m/2}$

Mises stress → $\frac{\sigma_e}{Y(\epsilon_e)}$
 Void volume fraction → f^*
 Hydrostatic stress → $\frac{3p}{2Y(\epsilon_e)}$

Void growth and nucleation $\dot{f} = (1-f)\dot{\epsilon}_m^p + N_v \dot{\epsilon}_e$

Void nucleation rate $N_v = \begin{cases} \frac{f_N}{s_N \sqrt{2\pi}} \exp\left(-\frac{1}{2} \left[\frac{\epsilon_m^p - \epsilon_N}{s_N} \right]^2\right) \dot{\epsilon}_e & p > 0 \\ 0 & p < 0 \end{cases}$

Void coalescence $f^* = \begin{cases} f & f < f_c \\ f_c + (1/q_1 - f_c)(f - f_c)/(f_F - f_c) & f \geq f_c \end{cases}$

Element deleted when $f = f_F$

Material Properties

Yield and hardening $Y(\epsilon_e)$

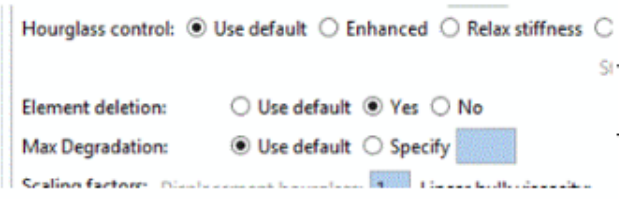
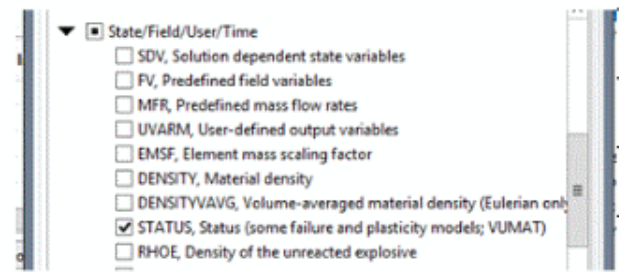
Void growth rate q_1, q_2

Void coalescence f_c, f_F

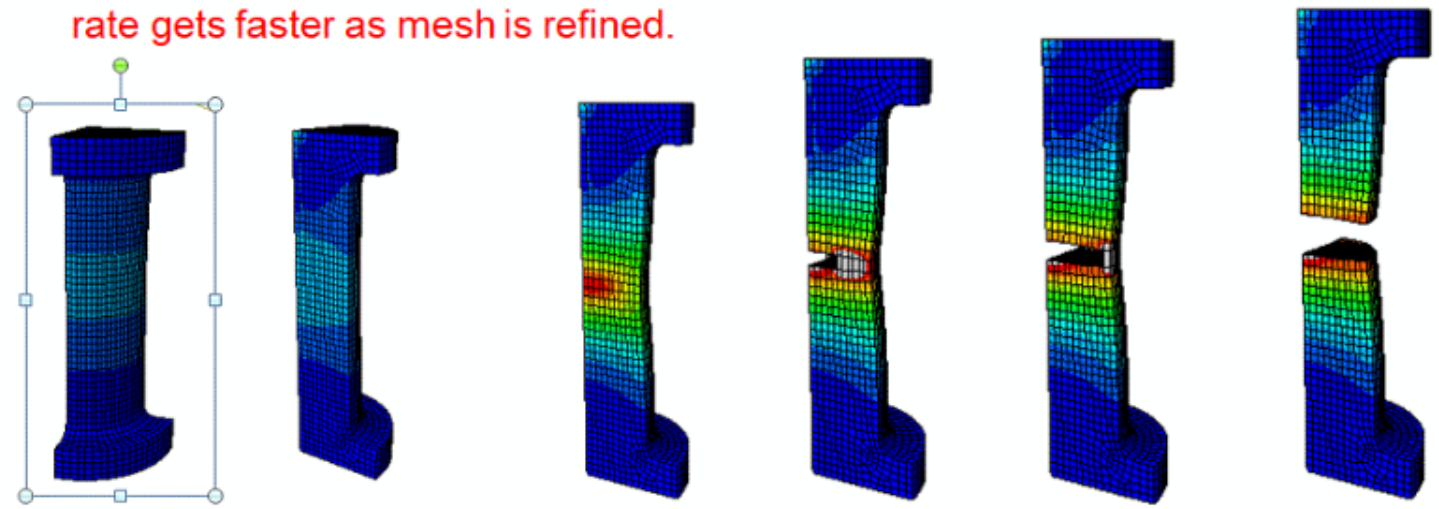
Void nucleation rate f_N, s_N, ϵ_N

Running ABAQUS/Explicit with Gurson Model

1. Set up geometry, section, etc in usual way.
2. In the Material module select Porous Metal Plasticity
3. Enter values for q1,q2,q3 (can use 1 for each)
4. Use the Suboptions button to define $f_N, s_N, \epsilon_N, f_C, f_F$
5. Define the initial value of $r=1-f$ in the 'Relative Density'
6. Create part instance in assembly in usual way
7. Create an explicit dynamic step. To enable element deletion, use Results->Field Output and in the dialog make sure the 'Status' option is checked
8. Apply boundary conditions in usual way
9. Mesh solid – in the Element Type menu check the box for 'Element Deletion'
10. Run job in usual way



11. **WARNING:** Simulations with models like this are always mesh sensitive once material starts to soften – softening rate gets faster as mesh is refined.



Ductile failure by necking & strain localization

In ductile materials failure can occur by material losing strength ; or geometric changes (or both)

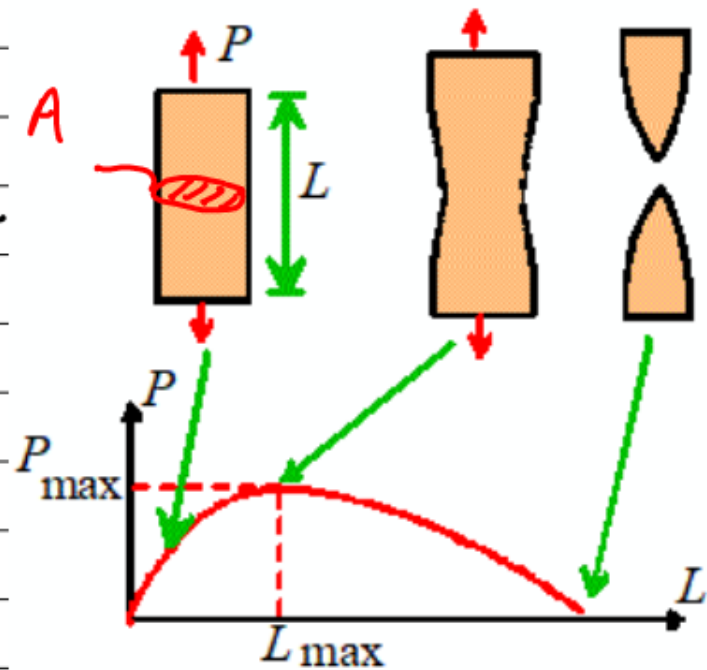
Predicting necking in tension

Predict P_{max} in uniaxial tension test

Assume material has true stress - ν - true strain curve

$$\sigma = Y(\epsilon) \quad \epsilon = \log(L/L_0)$$

Note $\frac{dP}{dL} = 0$ when $P = P_{max}$



Note also $P = A \sigma$

$$\frac{dP}{dL} = \frac{dA}{dL} \sigma + A \frac{d\sigma}{d\varepsilon} \frac{d\varepsilon}{dL} \quad \frac{d\varepsilon}{dL} = \frac{1}{L}$$

To find $\frac{dA}{dL}$ assume vol is constant $\Rightarrow AL = \text{const}$

$$\Rightarrow \frac{dA}{dL} L + A = 0 \quad \Rightarrow \frac{dA}{dL} = -\frac{A}{L}$$

$$\Rightarrow -\frac{A}{L} \sigma + \frac{A}{L} \frac{d\sigma}{d\varepsilon} = 0$$

At P_{\max}

$$\sigma = \frac{d\sigma}{d\varepsilon}$$

"Considere criterion"

Examples:

Linear hardening $\sigma = \sigma_0 + h \epsilon$

$$\Rightarrow \sigma_0 + h \epsilon_f = h \Rightarrow \epsilon_f = 1 - \frac{\sigma_0}{h}$$

Power-law $\sigma = \sigma_0 \left(\frac{\epsilon}{\epsilon_0} \right)^m$ $m < 1$

$$\frac{d\sigma}{d\epsilon} = \sigma \Rightarrow \frac{m \sigma_0}{\epsilon} \left(\frac{\epsilon}{\epsilon_0} \right)^m = \sigma_0 \left(\frac{\epsilon}{\epsilon_0} \right)^m$$

$\Rightarrow \epsilon_f = m$ at point of failure

High hardening, low flow stress makes material resist necking

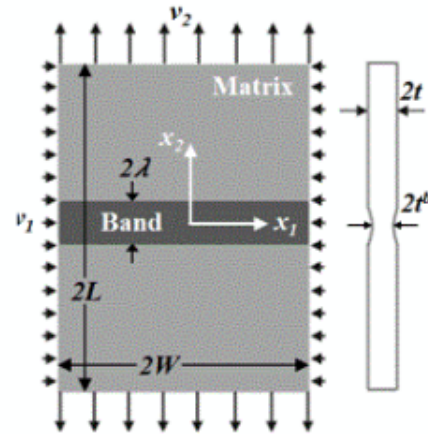
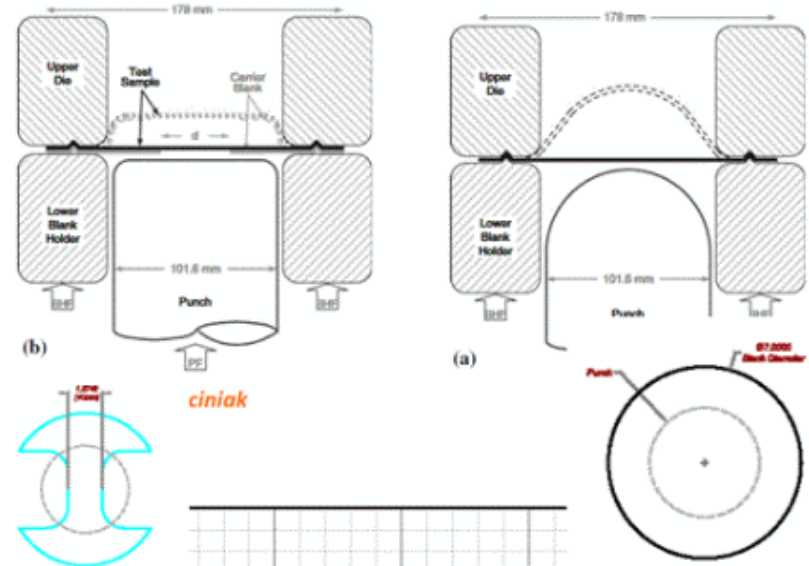
In general parts can fail by local thinning

Necking in sheet metals – Forming Limit Diagram

'Forming Limit Diagram' measured using both Marciniak and Nakajima experiments

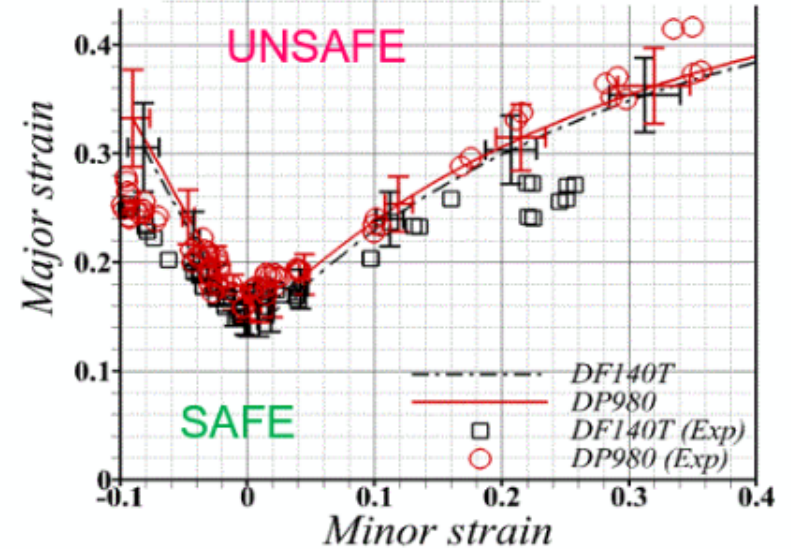
Predicted with 'M-K' analysis – calculate strain that causes necking in a thin sheet with a small defect

FLD can be defined in ABAQUS as a failure criterion



Major Strain ϵ_{22}

Minor Strain ϵ_{11}



14.3 Simple criteria for failure under cyclic loading

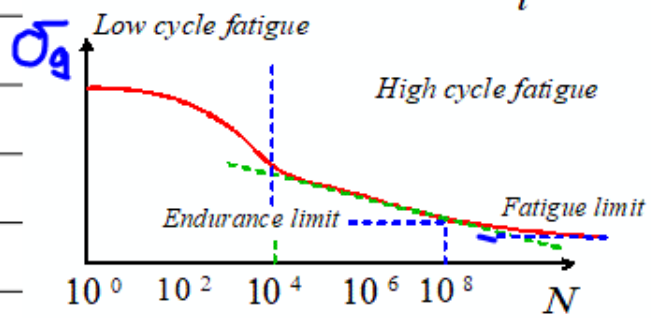
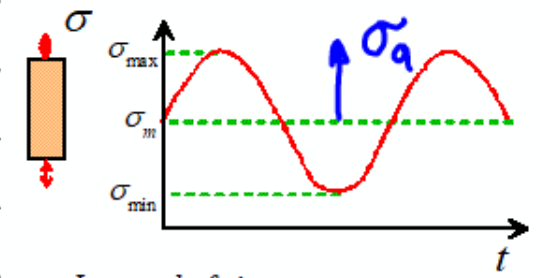
Typical behavior in cyclic tension:

(1) "Low Cycle" fatigue

$$N < 10^4$$

Yield stress exceeded

Plastic strain controlled



cycles to failure ↑

(2) "High Cycle" fatigue

$$N > 10^4$$

Below yield

Stress controlled

"Endurance Limit" : critical stress amplitude to survive 10^8 cycles

"Fatigue Limit" : critical stress " for ∞ life

Failure criterion for high cycle fatigue
 "Basquin's law"

$$N^b \bar{\sigma}_a = C \quad (b, C) \text{ material props}$$

where $\bar{\sigma}_a = (\sigma_{\max} - \sigma_{\min}) / 2$

C depends on mean stress $\bar{\sigma}_m = (\sigma_{\max} + \sigma_{\min}) / 2$

Goodman's rule $C = C_0 \left(1 - \frac{\bar{\sigma}_m}{\sigma_{UTS}} \right)$

Where C_0 is value of C for fully reversed loading
 $\bar{\sigma}_m = 0$

σ_{UTS} = Tensile strength

Low Cycle Fatigue criterion

Low cycle fatigue controlled by plastic strain amplitude

$$N^d \Delta \epsilon^p = D$$

where (d, D) are material properties

"Coffin - Manson" law.

