



School of Engineering
Brown University

EN2210: Continuum Mechanics

Midterm Examination Wed Oct 26 2016

NAME: _____

General Instructions

- No collaboration of any kind is permitted on this examination.
- You may bring 2 double sided pages of reference notes. No other material may be consulted
- Write all your solutions in the space provided. No sheets should be added to the exam.
- Make diagrams and sketches as clear as possible, and show all your derivations clearly. Incomplete solutions will receive only partial credit, even if the answer is correct.
- If you find you are unable to complete part of a question, proceed to the next part.

Please initial the statement below to show that you have read it

By affixing my name to this paper, I affirm that I have executed the examination in accordance with the Academic Honor Code of Brown University. **PLEASE WRITE YOUR NAME ABOVE ALSO!**

1 (5 points) _____

2. (5 points) _____

3. (5 points) _____

4. (5 points) _____

5. (8 points) _____

TOTAL (28 points) _____

1. Let \mathbf{R} be a proper orthogonal tensor satisfying $\mathbf{R}^T \mathbf{R} = \mathbf{R} \mathbf{R}^T = \mathbf{I}$, $\det(\mathbf{R})=1$, and let \mathbf{a}, \mathbf{b} be vectors. Using index notation, show that $(\mathbf{R}\mathbf{a}) \times (\mathbf{R}\mathbf{b}) = \mathbf{R}(\mathbf{a} \times \mathbf{b})$. You will need the following result for a determinant:

$$\epsilon_{lmn} \det(\mathbf{A}) = \epsilon_{ijk} A_{il} A_{jm} A_{kn}$$

$$\begin{aligned} (\mathbf{R}\mathbf{a}) \times (\mathbf{R}\mathbf{b}) &\equiv \epsilon_{ijk} R_{jn} a_n R_{km} b_m = \epsilon_{qjk} \delta_{iq} R_{jn} R_{km} a_n b_m = \epsilon_{qjk} R_{ql} R_{il} R_{jn} R_{km} a_n b_m \\ &= R_{il} \epsilon_{qjk} R_{ql} R_{jn} R_{km} = R_{il} \det(\mathbf{R}) \epsilon_{lmn} a_n b_m = R_{il} \epsilon_{lmn} a_n b_m = \mathbf{R}(\mathbf{a} \times \mathbf{b}) \end{aligned}$$

(5 POINTS)

2. Let \mathbf{a}, \mathbf{b} be two unit vectors. Find expressions for the eigenvalues and eigenvectors of $\mathbf{S} = \mathbf{a} \otimes \mathbf{b} + \mathbf{b} \otimes \mathbf{a}$, in terms of \mathbf{a}, \mathbf{b} and $\mathbf{a} \cdot \mathbf{b}$

- Since $\mathbf{S}\mathbf{u}$ must always lie in the \mathbf{a}, \mathbf{b} plane, one eigenvector is $\mathbf{a} \times \mathbf{b}$ and the corresponding eigenvalue is zero.
- We can construct the other eigenvectors as $\mathbf{a} + \beta \mathbf{b}$ where β is to be found. The definition of an eigenvector/value pair yields

$$\mathbf{S}(\mathbf{a} + \beta \mathbf{b}) = (\mathbf{a} \otimes \mathbf{b} + \mathbf{b} \otimes \mathbf{a})(\mathbf{a} + \beta \mathbf{b}) = \mathbf{a}[(\mathbf{a} \cdot \mathbf{b}) + \beta] + \mathbf{b}[1 + \beta(\mathbf{a} \cdot \mathbf{b})] = \gamma(\mathbf{a} + \beta \mathbf{b})$$

- We can see by inspection that choosing $\beta = 1 \Rightarrow \mathbf{a}[(\mathbf{a} \cdot \mathbf{b}) + 1] + \mathbf{b}[1 + (\mathbf{a} \cdot \mathbf{b})] = [1 + (\mathbf{a} \cdot \mathbf{b})](\mathbf{a} + \mathbf{b})$ so $\gamma = 1 + \mathbf{a} \cdot \mathbf{b}$ is an eigenvalue and $(\mathbf{a} + \mathbf{b})$ is an eigenvector. The other eigenvector must be perpendicular to this one so $\pm(\mathbf{a} - \mathbf{b})$ is the other eigenvector

You can also work step-by-step through the algebra:

$$[\mathbf{a} \cdot \mathbf{b} + \beta] + (\mathbf{a} \cdot \mathbf{b})[1 + \beta(\mathbf{a} \cdot \mathbf{b})] = \gamma[1 + \beta(\mathbf{a} \cdot \mathbf{b})]$$

$$(\mathbf{a} \cdot \mathbf{b})[\mathbf{a} \cdot \mathbf{b} + \beta] + [1 + \beta(\mathbf{a} \cdot \mathbf{b})] = \gamma[\mathbf{a} \cdot \mathbf{b} + \beta]$$

$$\Rightarrow [\mathbf{a} \cdot \mathbf{b} + \beta]^2 + (\mathbf{a} \cdot \mathbf{b})[1 + \beta(\mathbf{a} \cdot \mathbf{b})][\mathbf{a} \cdot \mathbf{b} + \beta] - (\mathbf{a} \cdot \mathbf{b})[\mathbf{a} \cdot \mathbf{b} + \beta][1 + \beta(\mathbf{a} \cdot \mathbf{b})] - [1 + \beta(\mathbf{a} \cdot \mathbf{b})]^2 = 0$$

$$\Rightarrow \beta^2(1 - (\mathbf{a} \cdot \mathbf{b})^2) + (\mathbf{a} \cdot \mathbf{b})^2 - 1 = 0$$

$$\Rightarrow \beta = \pm 1$$

Thus

$$(\mathbf{a} \otimes \mathbf{b} + \mathbf{b} \otimes \mathbf{a})(\mathbf{a} + \mathbf{b}) = (\mathbf{a} \cdot \mathbf{b})\mathbf{a} + \mathbf{b} + \mathbf{a} + \mathbf{b}(\mathbf{a} \cdot \mathbf{b}) = [1 + (\mathbf{a} \cdot \mathbf{b})](\mathbf{a} + \mathbf{b})$$

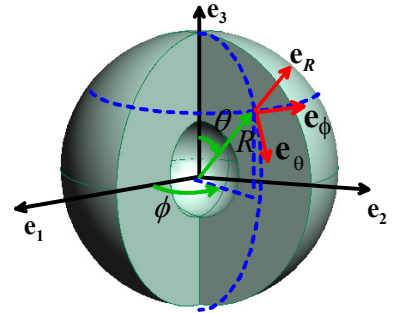
$$(\mathbf{a} \otimes \mathbf{b} + \mathbf{b} \otimes \mathbf{a})(\mathbf{a} - \mathbf{b}) = (\mathbf{a} \cdot \mathbf{b})\mathbf{a} + \mathbf{b} - \mathbf{a} - \mathbf{b}(\mathbf{a} \cdot \mathbf{b}) = -[1 - (\mathbf{a} \cdot \mathbf{b})](\mathbf{a} - \mathbf{b})$$

The eigenvalues are therefore $[1 + (\mathbf{a} \cdot \mathbf{b})]$, $-[1 - (\mathbf{a} \cdot \mathbf{b})]$

(5 POINTS)

3. The figure shows an incompressible spherical shell with inner radius a and outer radius b . The interior is being inflated at constant rate $da/dt = \alpha$.

3.1 Assume that the velocity field is radial $\mathbf{v} = v_R(r)\mathbf{e}_R$. Calculate an expression for the velocity gradient in terms of v_R , and hence show that $v_R = \alpha a^2 / R^2$.



$$\mathbf{L} = \nabla_{\mathbf{y}} \mathbf{v} = \frac{\partial v_R}{\partial R} \mathbf{e}_R \otimes \mathbf{e}_R + \frac{v_R}{R} (\mathbf{e}_\theta \otimes \mathbf{e}_\theta + \mathbf{e}_\phi \otimes \mathbf{e}_\phi)$$

Incompressibility requires $\text{tr}(\mathbf{L})=0$, which yields

$$\frac{dv_R}{dR} = -\frac{2v_R}{R} \Rightarrow \log(v_R) = -2\log(R) \Rightarrow v_R = \alpha \frac{a^2}{R^2}$$

[3 POINTS]

3.2 Suppose that a fly walks around the outer surface of the shell with constant speed w relative to the surface. Find a formula for the acceleration of the fly.

The acceleration is

$$\begin{aligned} \mathbf{a} &= \left. \frac{\partial \mathbf{v}}{\partial t} \right|_{\mathbf{y}} + (\nabla_{\mathbf{y}} \mathbf{v}) \mathbf{v} = 2\alpha^2 \frac{a}{R^2} \mathbf{e}_R + \alpha \left(-\frac{a^2}{R^3} \mathbf{e}_R \otimes \mathbf{e}_R + \frac{a^2}{R^3} (\mathbf{e}_\theta \otimes \mathbf{e}_\theta + \mathbf{e}_\phi \otimes \mathbf{e}_\phi) \right) \left(\alpha \frac{a^2}{R^2} \mathbf{e}_R + w(\beta \mathbf{e}_\theta \pm \sqrt{1-\beta^2} \mathbf{e}_\phi) \right) \\ &= \left(2\alpha^2 \frac{a}{R^2} - \alpha^2 \frac{a^4}{R^5} \right) \mathbf{e}_R + 2\alpha w \frac{a^2}{R^3} (\beta \mathbf{e}_\theta \pm \sqrt{1-\beta^2} \mathbf{e}_\phi) \quad -1 < \beta < 1 \end{aligned}$$

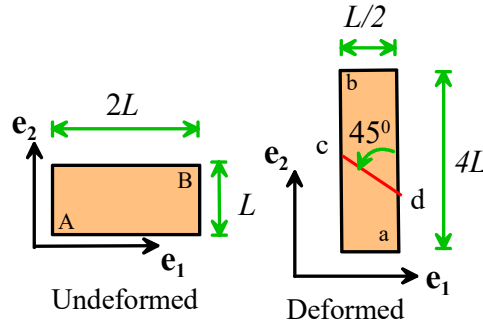
[2 POINTS]

4. The figure shows the reference and deformed configurations for a solid. The corners A and B map to a and b after deformation. Material fibers perpendicular to the plane retain their length and orientation during deformation.

The stress-strain law for the material relates the material stress Σ to the Lagrange strain \mathbf{E} by

$$\Sigma = 4\mathbf{E}$$

Calculate the traction acting on the plane c-d shown in the figure.



- Recall $\mathbf{F}=\mathbf{VR}$

$$\mathbf{J}\mathbf{F}^{-1}\boldsymbol{\sigma}\mathbf{F}^{-T} = 2(\mathbf{F}^T\mathbf{F} - \mathbf{I}) \Rightarrow \boldsymbol{\sigma} = \frac{1}{J}(\mathbf{F}\mathbf{F}^T\mathbf{F}\mathbf{F}^T - \mathbf{F}\mathbf{F}^T) = \frac{2}{J}\left(\left(\mathbf{V}\mathbf{R}\mathbf{R}^T\mathbf{V}\right)^2 - \mathbf{V}\mathbf{R}\mathbf{R}^T\mathbf{V}\right) = \frac{2}{J}(\mathbf{V}^4 - \mathbf{V}^2)$$

$$\boldsymbol{\sigma} = 2\left(\begin{bmatrix} 1/2 & 0 \\ 0 & 2 \end{bmatrix}^4 - \begin{bmatrix} 1/2 & 0 \\ 0 & 2 \end{bmatrix}^2\right) = \begin{bmatrix} -3/8 & 0 \\ 0 & 24 \end{bmatrix}$$

- The traction follows as

$$\mathbf{n} \cdot \boldsymbol{\sigma} = \frac{1}{\sqrt{2}} \begin{bmatrix} -3/8 \\ 24 \end{bmatrix}$$

- Alternatively, the deformation gradient can be constructed as a sequence of stretch and rotation

$$\mathbf{F} = \begin{bmatrix} 0 & -1 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} 2 & 0 \\ 0 & 1/2 \end{bmatrix} = \begin{bmatrix} 0 & -1/2 \\ 2 & 0 \end{bmatrix}$$

- The Lagrange strain follows as

$$\mathbf{E} = \frac{1}{2}(\mathbf{F}^T\mathbf{F} - \mathbf{I}) = \frac{1}{2}\left(\begin{bmatrix} 0 & 2 \\ -1/2 & 0 \end{bmatrix} \begin{bmatrix} 0 & -1/2 \\ 2 & 0 \end{bmatrix} - \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}\right) = \frac{1}{2} \begin{bmatrix} 3 & 0 \\ 0 & -3/4 \end{bmatrix}$$

(Can also get this as $(\mathbf{U}^2 - \mathbf{I})/2$)

- The material stress follows as

$$\Sigma = 2(\mathbf{F}^T\mathbf{F} - \mathbf{I}) = \begin{bmatrix} 6 & 0 \\ 0 & -3/2 \end{bmatrix}$$

- The Cauchy stress is then

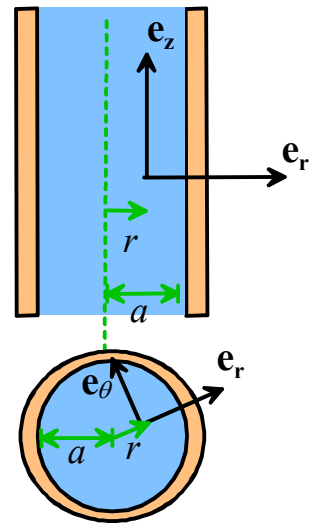
$$\boldsymbol{\sigma} = \mathbf{J}\mathbf{F}\Sigma\mathbf{F}^T = \begin{bmatrix} 0 & -1/2 \\ 2 & 0 \end{bmatrix} \begin{bmatrix} 6 & 0 \\ 0 & -3/2 \end{bmatrix} \begin{bmatrix} 0 & 2 \\ -1/2 & 0 \end{bmatrix} = \begin{bmatrix} 0 & -1/2 \\ 2 & 0 \end{bmatrix} \begin{bmatrix} 0 & 12 \\ 3/4 & 0 \end{bmatrix} = \begin{bmatrix} -3/8 & 0 \\ 0 & 24 \end{bmatrix}$$

[5 POINTS]

5. The figure shows an infinitely long cylindrical tube with a vertical axis. A fluid with mass density ρ and viscosity μ flows down the tube with steady velocity $\mathbf{v} = v(r)\mathbf{e}_z$ under the action of an axial gravitational force $\mathbf{b} = -\rho g\mathbf{e}_z$. The Cauchy stress in the fluid is

$$\boldsymbol{\sigma} = 2\mu\mathbf{D}$$

where \mathbf{D} is the stretch rate.



[2 POINTS]

5.1 Find an expression for the velocity gradient for the flow using the cylindrical-polar coordinate system shown in the figure.

$$\mathbf{L} = \nabla\mathbf{v} = v(r)\mathbf{e}_z \otimes \left(\frac{\partial}{\partial r}\mathbf{e}_r + \frac{1}{r}\frac{\partial}{\partial\theta}\mathbf{e}_\theta + \frac{\partial}{\partial z}\mathbf{e}_z \right) = \frac{\partial v}{\partial r}\mathbf{e}_z \otimes \mathbf{e}_r$$

5.2 Using the principle of virtual work, show that the velocity distribution must satisfy

$$\frac{1}{r}\frac{\partial}{\partial r}\left(r\frac{\partial v}{\partial r}\right) = \frac{\rho g}{\mu}$$

[4 POINTS]

$$\delta P = \int_V 2\mu\mathbf{D} : \delta\mathbf{D} - \int_V \mathbf{b} \cdot \delta\mathbf{v} dV = \int_0^a \mu \frac{\partial v}{\partial r} \frac{\partial \delta v}{\partial r} 2\pi r dr + \int_0^a \rho g \delta v 2\pi r dr = 0$$

Integrate by parts:

$$\begin{aligned} \int_0^a \mu \frac{\partial v}{\partial r} \frac{\partial \delta v}{\partial r} 2\pi r dr &= \int_0^a \mu \left[\frac{\partial}{\partial r} \left(\frac{\partial v}{\partial r} r \delta v \right) - \delta v \frac{\partial}{\partial r} \left(\frac{\partial v}{\partial r} r \right) \right] 2\pi dr \\ &= 2\pi\mu \left(\frac{\partial v}{\partial r} b \delta v(b) - \frac{\partial v}{\partial r} a \delta v(a) \right) - \int_0^a \mu \delta v \left(\frac{\partial^2 v}{\partial r^2} + \frac{1}{r} \frac{\partial v}{\partial r} \right) 2\pi r dr \end{aligned}$$

Substitute back into PVW

$$2\pi\mu \left(\frac{\partial v}{\partial r} b \delta v(b) - \frac{\partial v}{\partial r} a \delta v(a) \right) - \int_a^b \mu \delta v \left(\frac{\partial^2 v}{\partial r^2} + \frac{1}{r} \frac{\partial v}{\partial r} - \frac{\rho g}{\mu} \right) 2\pi r dr = 0$$

The integrand must vanish which yields the answer.

5.3. Calculate $v(r)$

$$\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial v}{\partial r} \right) = \frac{\rho g}{\mu} \Rightarrow r \frac{\partial v}{\partial r} = \frac{1}{2} \frac{\rho g}{\mu} r^2 + C \Rightarrow \frac{\partial v}{\partial r} = \frac{1}{2} \frac{\rho g}{\mu} r + \frac{C}{r}$$
$$\Rightarrow v = \frac{1}{4} \frac{\rho g}{\mu} r^2 + C \log(r) + D$$

The velocity must be bounded at $r=0$ so $C=0$, and $v=0$ at $r=a$ whence $v = \frac{\rho g}{4\mu} (r^2 - a^2)$

[2 POINTS]