ENGN 2340 Final Project: Viscoelasticity in soft biological tissues

Xuliang Qian

Many soft biological tissues have viscoelastic effects.

In 1959, Harkness and Harkness measured creep response of rat cervical tissue subjected to a constant dilating force experimentally. Below is a modified figure borrowed from ENGN 2220.



We can see that the properties of rat cervical tissue experience a dramatic transformation during pregnancy, in particular, the creep goes almost linear with time during childbirth. This is in accordance to its function – to help the passage of the child.

In ENGN 2220, we modeled this viscoelastic behavior using the following rheological model (the Burgers model), which consists of a Maxwell element in series with a Kelvin-Voigt element:



In fact, the Burgers model could also capture the creep for a viscoelastic liquid. For simplicity, we just took the stiffness of both springs to be E and the viscosity of both dashpots to be η .

The derivation for 1D governing equation is:

For Maxwell element J- T- m, En 4, 5, We have: $O_{M} = E_{i} \varepsilon_{M}^{e} = \frac{y_{i}}{\varepsilon_{M}} \varepsilon_{M}^{e}$ $\int \varepsilon_{M} = \varepsilon_{M}^{e} + \varepsilon_{M}^{e}$ $\Rightarrow \mathcal{E}_{M} = \mathcal{E}_{M}^{2} + \mathcal{E}_{M}^{2} = \frac{\mathcal{O}_{M}}{\mathcal{E}_{i}} + \frac{\mathcal{O}_{H}}{\mathcal{V}_{i}}$ 2) for Kelvin-Voigt element: J-II- OV, EV we have: Or = Ezer + 42er 5) For Burgers model : Maxwell in series with Kelvirtbigt: $\int \overline{O} = \overline{O} \overline{M} = \overline{O} \overline{V}$ $\int \frac{1}{2} = \underline{S} \underline{M} + \underline{S} \underline{V}$ combre: 0= 52(2- Ey) + 1/2 (2-Ey) $=E_{2}(\hat{z}-\frac{\delta}{E_{1}}-\frac{\delta}{y_{1}})+y_{2}(\hat{z}-\frac{\delta}{E_{1}}-\frac{\delta}{y_{1}}),$ $\Rightarrow \sigma + \left(\frac{\mu}{E} + \frac{\mu}{E} + \frac{\mu}{E^2}\right)\sigma + \frac{\mu}{EE}\sigma = \eta \varepsilon + \frac{\eta}{E}\varepsilon$ $\left[1 + \left(\frac{y_{i}}{E_{2}} + \frac{y_{i}}{E_{1}} + \frac{y_{z}}{E_{2}}\right)\frac{d}{dt} + \frac{y_{i}y_{z}}{E_{2}}\frac{d^{2}}{dt^{2}}\right]\delta(t) = \left(y_{i}\frac{d}{dt} + \frac{y_{i}y_{z}}{E_{2}}\frac{d^{2}}{dt^{2}}\right)E(t)$ Governing Equation for iD. if assume EI=Ez, yI=nz $= \left[1 + \frac{3N}{E} \frac{d}{dt} + \left(\frac{M}{E} \frac{2d^2}{dt} \right] \sigma(t) = \left(\frac{M}{dt} + \frac{M}{E} \frac{d^2}{dt} \right) \epsilon(t)$

For simplified 1D equation (E1=E2, η 1= η 2), we could show that the strain and creep for a stress step input is:

$$C(t) = \frac{1}{E} [1 - exp(-\frac{E}{y}t)]\sigma_0 + (\frac{1}{E} + \frac{1}{y})\sigma_0$$

$$J_{C}(t) = \frac{C(t)}{\sigma_0} = \frac{1}{E} [1 - exp(-\frac{E}{y}t)] + (\frac{1}{E} + \frac{1}{y}).$$

By plotting E Jc(t) versus $(E/\eta)t$, we could see the behavior describe earlier:



Next step is to generalize 1D to 3D. The derivation for 3D governing equation is shown below:

For 3D: E> Cijke Y> Cijke (Sty + 3 Cijre (CE) Kergart Cijmu Cmnrs (CE)rsxy (CE)xypean $= \left[\begin{array}{c} C_{ij} p_{g} \frac{d}{dt} + C_{ijmn} \begin{array}{c} C_{mn} rs \left(C^{E} \right) + sp_{g} \frac{d^{2}}{dt^{2}} \end{array} \right]$ Governing Equation for 3D E_1 0000000 However, it could be rather hard to implement. E_{2} ŋ

So I switch to a simpler but well-known model, the standard linear solid (SLS) model, see above, and successfully implement the SLS model in ABAQUS UMAT.

The pages below show the derivation of the theoretical side of FEM implementation.

ENGN 2340 Final Project : Theoretical Part: by Xuliang Riam We are studying FEM implement of SLS model. Cinstead of Burgers model) for simplicity. From under graduate textbook, the SLS model is defined as: A The Tel EI, E2: springs with stiffness The Tel Tel Tel E2 The Tel Tel Tel Descrity Main Main Main Mark Miscosity Main Main Main Mark Miscosity D Simple spring demant : A-min->. Ospring = Ei Espring Combine (0= Ospring + Oinaxusel) Of O (E = Espring = Emaxusel) $\implies \mathcal{E} = \frac{\sigma - \sigma_{\text{spring}}}{E_2} + \frac{\sigma - \sigma_{\text{spring}}}{\gamma_2}$ $\Rightarrow \mathcal{E} = \left(\frac{\mathcal{E}}{\mathcal{E}_{2}} - \frac{\mathcal{E}_{1}}{\mathcal{E}_{3}} \mathcal{E}\right) + \left(\frac{\mathcal{E}}{\mathcal{H}_{2}} - \frac{\mathcal{E}_{1}}{\mathcal{H}_{2}} \mathcal{E}\right)$ where "." stand for d/dt. $\Rightarrow (1 + \frac{E_1}{E_2}) = + \frac{E_1}{\eta_2} = \frac{\sigma}{E_2} + \frac{\sigma}{\eta_2} (For 1D)$ LHS have strain terms, RHS have stress terms, They are separated, which is always good! P1/10

sports a strange a spiral and an and the start Next, is to try something dangerous (but fun): To generalized this ID Governing Quartien into 3D! For 3D, we redefine the SLS model as follows: Ke, Gef Fre, Gre - 146, 45-1 In short, what I did here is : Ei -> Ke, Ge where: [Ke, Kye are bulk modulus $E_2 \rightarrow K_{Me}, G_{Me}$ Ge, Gue are shar modulus (Mb, Ms are Volume Viscosity 12 -> 1/6, 1/s. and shear viscosity respectively Revorite our governing equation in 3D form. First, look or 21ts. $LHS = (1 + \frac{G_{e}}{G_{Me}}) L \epsilon_{ij} - \frac{1}{3} S_{ij} tr(\epsilon_{i})] + \frac{1}{3} (1 + \frac{k_{e}}{k_{Me}}) S_{ij} tr(\epsilon_{i})$ $+\frac{Ge}{\eta_s} \left[\Sigma_{ij} - \frac{1}{3} S_{ij} tr(\Sigma) \right] + \frac{Ke}{\eta_s} \frac{1}{3} S_{ij} tr(\Sigma).$ Here I intentionally separate the volumetric strain and stear strain. This would make latter minupakition somewhat simpler. Then we take a look at RHS, it has the same dimension as (strain), but contains all the stress terms. Recall in ENGN 2210, $\underline{e} = \frac{1}{2G} \left[\underline{\sigma} - \frac{1}{2G+3\lambda} (\underline{tr} \underline{\sigma}) \right]$

plug in : $\lambda = K - \frac{2}{3}G$ $\Rightarrow \cdot \mathbf{G} = \frac{1}{2\mathbf{G}}\mathbf{G} + \frac{1}{3}\left(\frac{1}{3\mathbf{k}} - \frac{1}{2\mathbf{G}}\right) + \mathbf{r}(\mathbf{G})\mathbf{I}$ > RHS = -261ma 01 + 3 (3kine - 26me) + (2) Sig Combine & simplify: StrainTerm = StressTerm (where: StrainTerm = (H Ge) Eig + = (Ke - Ge) Sigtr(0). + Ge zig + - '(Ke - Ge) Sig tr(E). Stress Term ZGMO O' + 3 (3Kme ZGMe)+r(J)Sig $+\frac{1}{2\eta_s} \frac{1}{\eta_s} + \frac{1}{3} \left(\frac{1}{3\eta_s} - \frac{1}{2\eta_s} \right) tr(0) \delta_{\eta_s}$ So how to implement about into code (ABAQUE UMAT)? First we take good care of all the annoying perameters. Define: $\frac{1}{3}(\frac{1}{y_b} - \frac{1}{y_s}) = \lambda_1 \frac{1}{3}(\frac{1}{y_b} - \frac{1}{y_s}) = \lambda_2$ De = 2M1, (1+ Gre) = 2M2 $\Rightarrow \text{strainterm} = 2 \mathcal{U}_1 \mathcal{E}_{ij} + \lambda_1 \text{tr}(\underline{\varepsilon}) \mathcal{S}_{ij} \\ + 2 \mathcal{U}_2 \mathcal{E}_{ij} + \lambda_2 \text{tr}(\underline{\varepsilon}) \mathcal{S}_{ij}$ P\$/10

Also define: $\frac{1}{3}\left(\frac{1}{3y_b}-\frac{1}{3y_s}\right)\neq \varphi_i, \neq \left(\frac{1}{3t_{me}}-\frac{1}{2g_{me}}\right)=\varphi_2.$ = 24, , Zome = 2412 =?. Stress Term = 4, tr(5) Sig + 24, Oij + 1/2 tr (5) 8ig + 24/2 5ij Next, lets consider time discretization: Similar to what we did in class Oig > Oig + SOig Eig > Eig + <> Eig $+r(\underline{\varepsilon}) \rightarrow tr(\underline{\varepsilon}) + \Delta tr(\underline{\varepsilon}) + r(\underline{\varepsilon}) \rightarrow tr(\underline{\varepsilon}) + \sigma tr(\underline{\varepsilon})$ Eig > dzing Oij > di trig) -> dtrie) $+r(\underline{z}) \rightarrow \underline{dtr(\underline{z})}$ as $t \rightarrow t + 0b$. phy in and Simplify : E I double checked this part, it's correct! Strain Term . St $= 2\mathcal{M}, (\mathcal{E}_{ij} + \partial \mathcal{E}_{ij}) \partial t + \lambda, \delta_{ij} [t_r(\underline{s}) + \partial t_r(\underline{s})] \partial t$ $+ 2\mu_2 \partial E_{ij} + \lambda_2 \delta_{ij} \Delta tr(E).$ = $2 \text{ at } \text{ Eig} \cdot \mu_1 + 2 \text{ a } \text{ Eig} (\text{ at } \mu_1 + \mu_2)$ + $\text{ at } \text{ Sig} + r(\underline{\epsilon}) \cdot \lambda_1 + \text{ atr}(\underline{\epsilon}) \text{ Sig} (\text{ at } \lambda_1 + \lambda_2).$ P4/10

e di al proves Define: ablith=A the new later the start of the ot u, + u==B > strainTerm. at = Aatr(E) Sig + Ba Eig + at Gig tr(E) .], + 2at Eig . ll, Stress Term. St = 3t 4, Sig [tro) + stro)] + 2 st 4, (oig + 00ig). . . je. + P2 Sig str (2) + 24/2 00ij $= \Delta t \left[(p, tr(0) S_{ij} + 24, \sigma_{ij}) + (4, \Delta t + 4_2) \Delta tr(0) S_{ij} + 2(4, \Delta t + 4_2) \Delta \sigma_{ij} + 2(4, \Delta t + 4_2) \Delta \sigma_{ij} \right]$ Define: Ut 4, + 4=D St 4,+4=F => StressTerm. St = DSijutr(0) + FOOin + a+ (P, + r (0) Sig + 24, Oig) Next, we separate normal stress and shear stress. To obtain normal components, just let i=) and Permite: $3 \text{ stress Term} \cdot Ot = 3 \text{ Dotr}(\underline{\sigma}) + Fotr(\underline{\sigma}),$ $+ dt [3 \varphi, tr(\varphi) + 24; tr(\varphi)].$ 3 strain Term. of = 3A str (E) + Botr (E) $+ dt [3], tr(2) + 2M_i tr(2)$ P5/10

Simplify; > 3. Stress Term. dt = (3D+F) + (0) + (39, +24,) + (0) $= (39, +24,) at + (39_2 + 24_3) atr(0)$ + at (39,+244) tr (0) $Define: G = 5, at + 5, [5_1 = 34] + 241, = 341, [5_2 = 34] + 241, = 341, [5_2 = 34] + 241, = 341, [5_2 = 34] + 241, = 341, [5_2 = 34], = 341$ Similarly: 3. Strain Term. 21t = $(3A+B)atr(s)+at(3),+2\mu,)tr(s)$. $= \left[(3\lambda_1 + 2\mu_1) \delta t + (3\lambda_2 + 2\mu_3) \right] \Delta tr(\underline{s})$ $+ ot(31, +2, u_1) tr(S)$ $\frac{\text{Define}: H = \omega_0 \text{st} + \omega_2}{(\omega_2 = 3\lambda_1 + 3\mu_1 = \frac{k_e}{\gamma_b})}$ $\frac{\omega_2 = 3\lambda_2 + 2\mu_2 = 1 + \frac{\kappa_2}{16}}{16}$ $\frac{1}{12} = \frac{1}{12} + \frac{\kappa_2}{16} = 1 + \frac{\kappa_2}{16}$ Compine equation (I) and (II), we have the reption: Gotr(c)+At.S, ·tr(c) = Hotr(E) + St. W, tr(E). $(\underline{\mathbb{T}}) - - = > \operatorname{str}(\underline{\sigma}) = -\frac{1}{G} \left[H \cdot \operatorname{tr}(\underline{\varepsilon}) + \operatorname{str}(\underline{w}, \operatorname{tr}(\underline{\varepsilon}) - \underline{\gamma}, \operatorname{tr}(\underline{\sigma}) \right]$ with the obtain equation (IC!, us are now able to cleangle Oij from stress Term. Pb/60

an . Let's take a look at XX component. i=j=X, Sij=1, Eij=Sxx, Eij=Dxx tr(E)=Oxx+Oyy+Ozz, tr(E)=Sxx+Eyy+Ezz.plug in eg.(II) stress tem. at. $= F \Delta \sigma_j + \frac{P}{G} [H \cdot \delta tr(\underline{s}) + \Delta t(\underline{w}, tr(\underline{s}) - \underline{s}, tr(\underline{s})] Sij$ + at [9, tr (2) Sig + 24, 0ig] $= FaG_{ij} + \frac{DH}{G} atr(\underline{z}) + at(\underline{\varphi}_{i} - \frac{DS_{i}}{G}) tr(\underline{z})S_{ij}$ plugin + st. 24 org + st - Bus, tree) Sig $i=j=x = Fall x + \frac{DH}{B} Eag x + agy + agg = 1 + at(4, -\frac{D}{G})(bx + ay + agg) + at(4, -\frac{D}{G})(bx + agg) + at(4, -$ Strain Farm. St - A str(E) big + Bazig + at Sig · tr(E)), + 20t Eig M, Plugin i=j=x-= $A(a\xi_{xx}+a\xi_{yy}+a\xi_{zz})$ + $Ba\xi_{xx}+at\lambda$, $(\xi_{xx}+\xi_{yy}+\xi_{zz})$ + 20t Exx (U, ---- (T) Combine. (IV) and (V) FOOXX. = (OEx+ DEYY+DERE) (A- - DI+)+DEXX.B + (Exx + Eyy + Ezz) (1, - Dio,) at + Exx : 24, at $-(0x+0y+0z)(q-\frac{ps}{q})dt - 0xx\cdot 24, dt$ 87/10

= DOX. = (AEx+dEyy+dEyy) × A-E F + dex y - F. + (Exx+ Eyy + See) x 26 x (), - D(0,) F (Six+ Eyy + See) x 26 x (), - B() + SXX x F x 241 + $(\overline{Ox} + \overline{Oyy} + \overline{OBR}) \times \frac{\delta f}{F} \times (- \overline{Y_i} + \frac{\overline{DS}}{\overline{q}})$ + $O_{XX} \times \frac{dt}{F} \times (-24,) - (VI)$ For other normal components, just change i= j= { 2. For shear components, it's much enseler! i=j, Sij= 0. Strain Term. 2t = FOOig + 205. 4 Oig Strain Term. 2t = Bes Eig + 205: 11 Eig combine: Doin = FIBOEin +20t(u, Ein - 41, Oin)) in ABAQUS UMAT, we use engineering shear strain Yig $E_{ij} = \frac{l_{ij}}{2}, \quad \Delta E_{ij} = \frac{\Delta l_{ij}}{2}$ => 210ig = == 27ig + = a. . Yig - == 4. . Oig 18/10

conte (VI) and (VII) in matrix form: (so that we could write this into UMAT). components Define: Stress Inc = [2044] Stress = [044 0088 - [082] Strain Inc = [d Exx] Strain = [Exy] UEEE] Strain = [Eyy] Define $EG = \begin{bmatrix} 1 & 1 & 1 \\ 1 & 1 & 1 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 1 \\ 0 & 0 & 1 \end{bmatrix}$ =>. Stress Inc = PDSIDDE · Strain Inc + DDSIDE · Strain (") + DDSIDE · Strain (") + DDSIDE · Strain (") where DDS DDE = A-B-EC] + BEI] $PDSPE'' = \underbrace{\Delta t}_{F}(I_{1} - \underbrace{Dw}_{G})[C] + \underbrace{Zt}_{F}(I_{1}CI)$ $PPSDS = -\frac{st}{F}(\varphi, -\frac{PS_i}{G})Ec_i - \frac{st}{F}\psi_i EI_j$ For shear components, we perform similar tricks: Define : Stress Inc = [doit] Stress = [0in] [dojk] Stress = [0ik] Strain Inc = $\begin{bmatrix} 0 & 2ij \\ 0 & 2ik \\ 0 & 2ik \\ 0 & 2ik \\ 0 & 2jk \end{bmatrix}$ Strain = $\begin{bmatrix} 2ij \\ 2ik \\ 2jk \end{bmatrix}$ > stress Inc = DDS DDE . Strain Inc + DPS DE + Strain (5) + DDSDS", Stress (S). P9 1.0

where: $DDSDDE = \frac{B}{2F} [I]$ $PPSDE = \frac{\Delta t}{F} \mathcal{U}, EE$ DDSDS = - 20+ 4, [I] Now that we save the time evolution but eforcedy, we implement this in code (ABRAUS UMAT) to see what happens. · is a triad to the second ن : الا الا الا الحيد 1. 1. 1 Pio/10

Results:

First, I test a simple 2-element model with prescribed displacement (time-dependant) at one end, fix the other end and let the rest faces be free. The prescribed displacement would increase to 0.1 from t=0 to t=10. After that, it will stay fixed.



The loading curve as well as the relaxation curve is in good agreement pure theoretical calculation (in 1D). This shows that the way I implement SLS model into FEM is correct.

Then I test the same 2-element sample with force loading (also time-dependent) in one end, fix the other end while letting the rest faces be free. The prescribed loading would increase to 500 from t=0 to t=1. After that, it will stay fixed. This setting is to mimic the loading condition for rat cervical tissue during childbirth.

The result is:



Here, we could see after a sudden increase (which corresponds to the sudden stress increment), the strain level will increase almost linearly with time (meanwhile, the stress is constant). This strain-time relation is in good agreement with the experimental result.

Therefore, I conclude that SLS model is able to capture the linear creep response observed in rat cervical tissue during childbirth.

Link to my Github repo: https://github.com/monyxl/EN234_FEA Here below is my UMAT code:

```
1
1
     ABAQUS format user material subroutine for small strain hypoelastic material
1
L
      SUBROUTINE UMAT(STRESS, STATEV, DDSDDE, SSE, SPD, SCD,
     1 RPL,DDSDDT,DRPLDE,DRPLDT,
     2 STRAN, DSTRAN, TIME, DTIME, TEMP, DTEMP, PREDEF, DPRED, CMNAME,
     B NDI, NSHR, NTENS, NSTATV, PROPS, NPROPS, COORDS, DROT, PNEWDT,
     4 CELENT, DFGRD0, DFGRD1, NOEL, NPT, LAYER, KSPT, KSTEP, KINC)
!
      INCLUDE 'ABA PARAM.INC'
I
      CHARACTER*80 CMNAME
     DIMENSION STRESS(NTENS), STATEV(NSTATV),
     1 DDSDDE(NTENS,NTENS),DDSDDT(NTENS),DRPLDE(NTENS),
     2 STRAN(NTENS),DSTRAN(NTENS),TIME(2),PREDEF(1),DPRED(1),
     B PROPS(NPROPS), COORDS(3), DROT(3,3), DFGRD0(3,3), DFGRD1(3,3)
      double precision :: KE, GE, KMe, GMe, GV, KV
      double precision :: lambda1, lambda2, mu1, mu2
      double precision :: phi1, phi2, psi1, psi2
      double precision :: omega1, omega2, zeta1, zeta2
      double precision :: A, B, C, D, F, G, H
      double precision, dimension(NTENS,NTENS) :: ones, diag
      double precision, dimension(NTENS,NTENS) :: DDSDE
      double precision, dimension(NTENS,NTENS) :: DDSDS
      double precision, dimension(NTENS) :: SIGold
      integer :: j
       KE = PROPS(1)
       GE = PROPS(2)
       KMe = PROPS(3)
       GMe = PROPS(4)
       GV = PROPS(5)
       !KV = PROPS(6) infinity
       !lambda1 = (-GE/GV+KE/KV)/3.d0
       lambda1 = (-GE/GV+0.d0)/3.d0
       lambda2 = (-GE/GMe+KE/KMe)/3.d0
       mu1 = GE/(2.d0*GV)
       mu2 = (1.d0 + GE/GMe)/2.d0
       phi1 = -1.d0/(6.d0*GV)!+1.d0/(9.d0*KV)
       phi2 = -1.d0/(6.d0*GMe) + 1.d0/(9.d0*KMe)
       psi1 = 1.d0/(4.d0*GV)
       psi2 = 1.d0/(4.d0*GMe)
       !omega1 = KE/KV
       omega1 = 0.d0
       omega2 = 1.d0 + KE/KMe
       !zeta1 = 1.d0/(3.d0*KV)
       zeta1 = 0.d0
       zeta2 = 1.d0/(3.d0*KMe)
       !ABCDFGH
```

```
A = lambda1*DTIME + lambda2
  B = 2.d0*(mu1*DTIME + mu2)
  D = phi1*DTIME + phi2
  F = 2.d0*(psi1*DTIME + psi2)
  G = zeta1*DTIME + zeta2
  H = omega1*DTIME + omega2
  !Initialize
  DDSDDE = 0.d0
  DDSDE = 0.d0
  DDSDS = 0.d0
  SIGold = 0.d0
  ones = 0.d0
  diag = 0.d0
  !Define Tensor coefficients
  ones(1:NTENS,1:NTENS) = 1.d0
  diag(1:NTENS,1:NTENS) = 0.d0
  forall(K1=1:NTENS) diag(K1,K1) = 1.d0
  !Define DDSDDE, DDSDE, DDSDS related to Normal Stress
  DDSDDE(1:NDI,1:NDI) = (A-D*H/G)/F*ones(1:NDI,1:NDI)
!
                              + B/F*diag(1:NDI,1:NDI)
  DDSDE(1:NDI,1:NDI) = DTIME/F*(lambda1-omega1*D/G)*
!
            ones(1:NDI,1:NDI) + 2.d0*DTIME*mu1/F*diag(1:NDI,1:NDI)
  DDSDS(1:NDI,1:NDI) = -DTIME*(-phi1+D*zeta1/G)/F*
!
            ones(1:NDI,1:NDI) - 2.d0*DTIME*psi1/F*diag(1:NDI,1:NDI)
  !Define DDSDDE, DDSDE, DDSDS related to Shear Stress
  DDSDDE(NDI+1:NTENS,NDI+1:NTENS) = B/2.d0/F
!
                                 *diag(NDI+1:NTENS,NDI+1:NTENS)
  DDSDE(NDI+1:NTENS, NDI+1:NTENS) = DTIME*mu1/F
!
                                 *diag(NDI+1:NTENS,NDI+1:NTENS)
  DDSDS(NDI+1:NTENS,NDI+1:NTENS) = -2.d0*DTIME*psi1/F
!
                                  *diag(NDI+1:NTENS,NDI+1:NTENS)
  !Store present stress state ub array SIGold
  do i = 1,ntens
      SIGold(i) = stress(i)
  end do
  !Update stresses
  do i = 1,ntens
  do j = 1,ntens
     stress(i) = stress(i) + DDSDDE(i,j)*(DSTRAN(j))
                           + DDSDE(i,j)*(STRAN(j))
                           + DDSDS(i,j)*(SIGold(j))
  end do
  end do
  return
 RETURN
 END SUBROUTINE UMAT
```

For input files, I edited the one we used in HW6 for porous elastic material, and play with the loading conditions:

1. For displacement loading, I used:

```
%
      The BOUNDARY conditions key starts definition of BCs
      BOUNDARY CONDITIONS
%
        The HISTORY key defines a time history that can be applied to DOFs or distributed
loads
        HISTORY, dof_history
          0.d0, 0.d0
                                      % Each line gives a time value and then a function
value
         10.d0, 0.1d0
        END HISTORY
        HISTORY, dload_history
          0.d0, 0.d0
         1.d0, 500.d0
        END HISTORY
%
        The NODESET key defines a list of nodes
        NODESET, node1
           1
         END NODESET
        NODESET, left
           1, 4, 5, 8
        END NODESET
        NODESET, right
           9, 10, 12, 11
        END NODESET
        NODESET, side
           1, 2, 5, 6, 11, 9
        END NODESET
       The ELEMENTSET key defines a list of elements
%
       ELEMENTSET, end element
           2
       END ELEMENTSET
        The DEGREE OF FREEDOM key assigns values to nodal DOFs
%
%
        The syntax is node set name, DOF number, and either a value or a history name.
%
        DEGREES OF FREEDOM
           1, 3, VALUE, 0.d0
           side, 2, VALUE, 0.d0
           left, 1, VALUE, 0.d0
           right, 1, HISTORY, dof_history
        END DEGREES OF FREEDOM
```

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2. For force loading, I used:

% The BOUNDARY conditions key starts definition of BCs BOUNDARY CONDITIONS The HISTORY key defines a time history that can be applied to DOFs or distributed % loads HISTORY, dof history 0.d0, 0.d0 % Each line gives a time value and then a function value 10.d0, 0.1d0 END HISTORY HISTORY, dload_history 0.d0, 0.d0 1.d0, 500.d0 END HISTORY % The NODESET key defines a list of nodes NODESET, node1 1 END NODESET NODESET, left 1, 4, 5, 8 END NODESET NODESET, right 9, 10, 12, 11 END NODESET NODESET, side 1, 2, 5, 6, 11, 9 END NODESET The ELEMENTSET key defines a list of elements % ELEMENTSET, end_element 2 END ELEMENTSET The DEGREE OF FREEDOM key assigns values to nodal DOFs % % The syntax is node set name, DOF number, and either a value or a history name. % DEGREES OF FREEDOM 1, 3, VALUE, 0.d0 side, 2, VALUE, 0.d0 left, 1, VALUE, 0.d0 END DEGREES OF FREEDOM DISTRIBUTED LOADS end element, 4, NORMAL, dload history END DISTRIBUTED LOADS

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