

Review – Field Equations for Ideal Fluids

- Simplified form of Navier-Stokes (Euler equation)

$$-\frac{1}{\rho} \frac{\partial \pi_{eq}}{\partial y_i} + b_i = \frac{\partial v_i}{\partial t} \Big|_{y_i=const} + \frac{1}{2} \frac{\partial}{\partial y_i} (v_k v_k) + \epsilon_{ijk} \omega_j v_k$$

- Vorticity Transport Equation

$$\epsilon_{ijk} \frac{\partial}{\partial x_j} (b_k) + D_{ij} \omega_j - \frac{\partial v_k}{\partial y_k} \omega_i = \frac{\partial \omega_i}{\partial t} \Big|_{\mathbf{x}=const}$$

- Irrotational Flow: Assume $\omega_i = 0$ at time $t=0$, and that $\epsilon_{ijk} \frac{\partial}{\partial x_j} (b_k) = 0$.
Then $\omega_i = 0$ for all time – flow is said to be **irrotational**.

Bernoulli Equation:

- Assumptions :
- (1) Ideal fluid - $\psi(\rho)$ $\pi_{eq} = \rho^2 \frac{\partial \psi}{\partial \rho}$
 - (2) Body force can be derived from a potential $\underline{b} = -\nabla \phi$
 - (3) Steady flow : $\frac{d\underline{u}}{dt} \Big|_y = 0$

Define streamline

Curves that are everywhere tangent to velocity

(Path traced by material particles)



Streamline is a path traced out by a massless particle moving with the flow.

Velocity is tangent to streamline at every point.

Mass does not cross streamlines.

Define $H = \psi + \frac{\pi_{eq}}{\rho} + \frac{1}{2} |\underline{v}|^2 + \phi$

Bernoulli : $H = \text{constant along streamlines}$

For irrotational flow $H = \text{const everywhere}$

Proof : (1) $\frac{1}{\rho} \frac{\partial \pi_{eq}}{\partial y_i} = \frac{\partial}{\partial y_i} \left(\psi + \frac{\pi_{eq}}{\rho} \right)$

To see this $\frac{\partial \psi}{\partial y_i} = \frac{\partial v}{\partial \rho} \frac{\partial \rho}{\partial y_i}$ $\frac{\partial}{\partial y_i} \left(\frac{\pi_{eq}}{\rho} \right) = \frac{1}{\rho} \frac{\partial \pi_{eq}}{\partial y_i} - \frac{\pi_{eq}}{\rho^2} \frac{\partial \rho}{\partial y_i}$

Recall $\pi_{eq} = \rho^2 \frac{\partial \psi}{\partial \rho} \Rightarrow (1)$

Now N-S eq can be re-written

$$-\frac{\partial}{\partial y_i} \left(\psi + \frac{\pi \rho g r^2}{\rho} \right) - \frac{\partial \phi}{\partial y_i} = \frac{\partial}{\partial y_i} \frac{1}{2} \underline{V}^2 + \underline{\omega} \times \underline{V}$$

$$\Rightarrow -\frac{\partial H}{\partial y_i} = \underline{\omega} \times \underline{V} \quad (2)$$

Now consider curve tangent to V

Dot (2) on both sides with tangent

$$-\frac{dH}{ds} = 0 \quad \Rightarrow \quad H = \text{const along curve}$$

Irrrotational $\Rightarrow H = \text{const}$

Potential Flow

Assumptions:

- (1) Irrotational
- (2) Incompressible
- (3) Inviscid

$$(1) \Rightarrow \nabla_y \times \underline{v} = 0 \quad \Leftrightarrow \quad \underline{v} = \nabla_y \Omega \quad \Omega(y, t)$$

$\Omega = \text{scalar function}$

$$(2) \Rightarrow \nabla_y \cdot \underline{v} = 0 \Rightarrow \nabla_j \nabla_j \Omega = \nabla_y^2 \Omega = 0$$

This equation (with boundary conditions) is the governing eq for \underline{v}

Bernoulli gives pressure

$$\frac{p}{\rho} + \frac{1}{2} |\underline{v}|^2 + \phi + \frac{\partial \Omega}{\partial t} = f(t)$$

Incompressible Viscous Flow

$$N-S: \quad -\frac{1}{\rho} \frac{\partial p}{\partial y_i} + \frac{\eta}{\rho} \frac{\partial^2 v_i}{\partial y_j \partial y_j} + b_i = \frac{Dv_i}{Dt} \Big|_y + \frac{\partial}{\partial y_i} \left(\frac{1}{2} |\underline{v}|^2 \right) + \underline{\omega} \times \underline{v}$$

Taking curl gives vorticity transport eq

$$\nabla_y \times \underline{b} + \frac{\eta}{\rho} \nabla_y^2 \underline{\omega} + D\underline{\omega} = \frac{D\underline{\omega}}{Dt} \Big|_x$$

Viscous flows need not be irrotational

Stokes flows :

Assume $|V|^2 \ll |\nabla_y^2 V|$

Then can neglect nonlinear terms in velocity

$$-\frac{1}{\rho} \frac{\partial p}{\partial y_i} + \frac{\eta}{\rho} \frac{\partial^2 v_i}{\partial y_j \partial y_j} + b_i \approx \frac{\partial v_i}{\partial t} \Big|_y$$

Solutions to fluid mechanics problems

- (1) Control vol method
- (2) Potential Flow
- (3) Compressible ideal flows
- (4) Stokes flow

Review: from earlier in notes

Mass Conservation:
$$\frac{d}{dt} \int_R \rho dV + \int_B \rho \mathbf{v} \cdot \mathbf{n} dA = 0$$

Linear Momentum Balance
$$\int_B \mathbf{n} \cdot \boldsymbol{\sigma} dA + \int_R \rho \mathbf{b} dV = \frac{d}{dt} \int_R \rho \mathbf{v} dV + \int_B (\rho \mathbf{v}) \mathbf{v} \cdot \mathbf{n} dA$$

Angular Momentum Balance
$$\int_B \mathbf{y} \times (\mathbf{n} \cdot \boldsymbol{\sigma}) dA + \int_R \mathbf{y} \times (\rho \mathbf{b}) dV = \frac{d}{dt} \int_R \mathbf{y} \times \rho \mathbf{v} dV + \int_B (\mathbf{y} \times \rho \mathbf{v}) \mathbf{v} \cdot \mathbf{n} dA$$

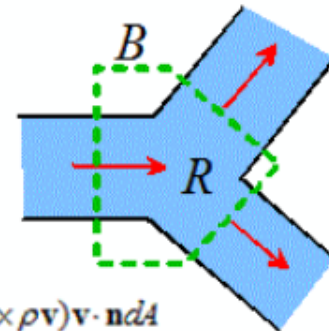
Mechanical Power Balance
$$\int_B (\mathbf{n} \cdot \boldsymbol{\sigma}) \cdot \mathbf{v} dA + \int_R \rho \mathbf{b} \cdot \mathbf{v} dV = \int_R \boldsymbol{\sigma} : \mathbf{D} dV + \frac{d}{dt} \int_R \frac{1}{2} \rho (\mathbf{v} \cdot \mathbf{v}) dV + \int_B \frac{1}{2} \rho (\mathbf{v} \cdot \mathbf{v}) \mathbf{v} \cdot \mathbf{n} dA$$

First law of thermodynamics

$$\int_B (\mathbf{n} \cdot \boldsymbol{\sigma}) \cdot \mathbf{v} dA + \int_R \rho \mathbf{b} \cdot \mathbf{v} dV - \int_B \mathbf{q} \cdot \mathbf{n} dA + \int_V \rho q dV = \frac{d}{dt} \int_R \rho \left(\varepsilon + \frac{1}{2} \mathbf{v} \cdot \mathbf{v} \right) dV + \int_B \rho \left(\varepsilon + \frac{1}{2} \mathbf{v} \cdot \mathbf{v} \right) \mathbf{v} \cdot \mathbf{n} dA$$

Second law of thermodynamics

$$\frac{d}{dt} \int_R \rho s dV + \int_B \rho s (\mathbf{v} \cdot \mathbf{n}) dA + \int_B \frac{\mathbf{q} \cdot \mathbf{n}}{\theta} dA - \int_R \frac{q}{\theta} dV \geq 0$$



Can use these to solve some problems directly

Assume ideal fluid

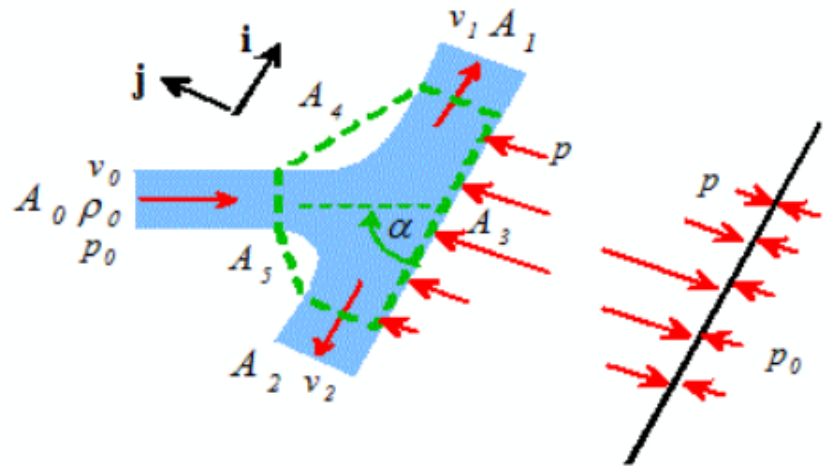
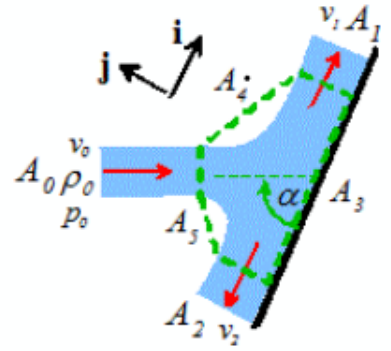
- no shear stress on wall

Force on wall is

$$-\int_{A_3} (p - p_0) dA_j = \underline{F}$$

Apply LMB to control vol shown

Example: Fluid jet hits an inclined wall. Calculate resultant force on the wall.



Linear Momentum:
$$\int_A \underline{n} \cdot \underline{\sigma} dA = \int_A (\rho \underline{v}) \underline{v} \cdot \underline{n} dA \quad (1)$$

On A_1, A_2, A_4, A_5 $\underline{\sigma} = \underbrace{-p_0 \underline{I}}_{\text{Atmospheric}}$ on A_3 $\underline{\sigma} = -p \underline{I}$

on A_2, A_1 $\underline{v} = \pm v \underline{i}$ on A_3 $\underline{v} \cdot \underline{n} = 0$

on A_0 $\underline{v} = v_0 \cos \alpha \underline{i} - v_0 \sin \alpha \underline{j}$ $\underline{n} = \underline{j}$

Hence (1) reduces to

$$\int_{A_1 + A_2 + A_4 + A_5} -p_0 \underline{n} + \int_{A_3} p \underline{j} dA = \int_{A_0} \rho v_0 (\cos \alpha \underline{i} - \sin \alpha \underline{j}) (-v_0 \sin \alpha) + \int_{A_2} \rho v_2 \underline{i} v_2 dA + \int_{A_1} \rho v_1 \underline{i} v_1 dA$$

Note $\int_{A_1+A_2+A_4+A_5} -p_0 \underline{n} = \int_{A_1+A_2+A_3+A_4+A_5} -p_0 \underline{n} dA + \int_{A_3} p_0 \underline{n} dA$

$\underline{n} = -\underline{j}$
on A_3

Note also $\int_{A_1+A_2+A_3+A_4+A_5} p_0 \underline{n} dA = 0$ (no force from uniform pressure on closed surface)

Hence $\int_{A_3} (p-p_0) \underline{j} dA = \rho v_0^2 \sin \alpha A_0 \underline{j} - \rho v_0^2 \cos \alpha A_0 \underline{i}$

$= -\underline{F}$

$+ \int_{A_{12}} \rho v_2 \underline{i} v_2 dA + \int_{A_1} \rho v_1 \underline{i} v_1 dA = 0$

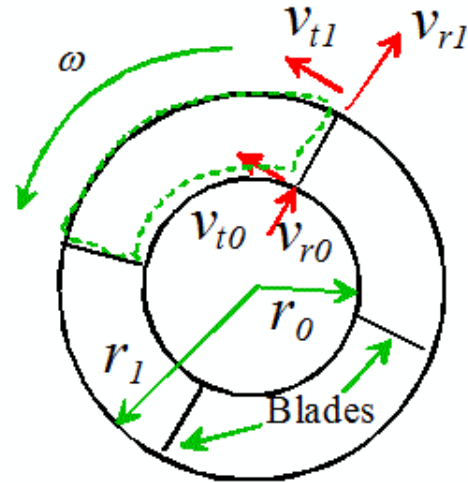
$$\underline{F} = -\rho v_0^2 \sin \alpha A_0 \underline{j}$$

Assumptions:

Ideal, incompressible fluid

Infinite number of vanes

Pump rotates with angular speed ω
The mass flow rate is μ
Estimate the torque on the shaft



volute centrifugal pump

<http://youtu.be/V3aPHmZ97yM>

