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7.0 Solving problems with nonlinear materials

Most common FEA coding problem

7.1 Hypoelasticity : small-strain nonlinear elastic material

Preliminaries : Governing equations

- Strains $\epsilon_{ij} = \text{sym}(\partial u_i / \partial x_j)$
- Equilibrium : weak form

$$\int_R \sigma_{ij} [\epsilon_{pq}(u_k)] \frac{\partial \eta_i}{\partial x_j} dV - \int_{S_2} \epsilon_i^* \eta_i dA = 0 \quad \forall \text{admiss } \eta_i$$

$$\sigma_{ij} n_j = \epsilon_i^* \text{ on } S_2$$

$$u_i = u_i^* \text{ on } S_1$$

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- Stress - strain law:

- (1) Reversible ; rate independent

- (2) Constant bulk modulus .

- (3) Uniaxial σ - ϵ curve nonlinear

$$\bar{\sigma}_{ij} = S_{ij} + \beta S_{ij} \quad \beta = k E_{kk}$$

$$\text{Let } \bar{\epsilon}_{ij} = \epsilon_{ij} - \frac{E_{kk}}{3} S_{ij} \quad E_e = \sqrt{\frac{2}{3} \bar{\epsilon}_{ij} \bar{\epsilon}_{ij}}$$

$$S_{ij} = \frac{2}{3} \bar{\sigma}_e (\epsilon_e) \frac{\bar{\epsilon}_{ij}}{\epsilon_p}$$

We can choose any convenient function for $\sigma_e(\epsilon_e)$

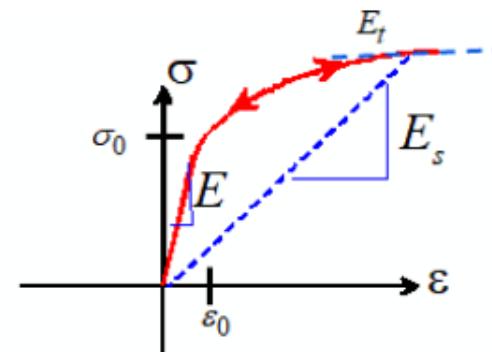
- ABAQUS uses a user-defined table

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ExampleMaterial properties:

$$\sigma_0, \varepsilon_0, n > 1$$

$$\frac{\sigma_e}{\sigma_0} = \begin{cases} \sqrt{\frac{1+n^2}{(n-1)^2} - \left(\frac{n}{n-1} - \frac{\varepsilon_e}{\varepsilon_0}\right)^2} - \frac{1}{n-1} & \varepsilon_e \leq \varepsilon_0 \\ \left(\frac{\varepsilon_e}{\varepsilon_0}\right)^{1/n} & \varepsilon_e \geq \varepsilon_0 \end{cases}$$



Power-law but w/ finite slope $\Rightarrow \varepsilon_e = 0$

FE equations : use usual FE interpolations

$$u_i = N^a u_i^a \quad \eta_i = N^a \eta_i^a$$

\Rightarrow PW:

$$\left\{ \int_R \sigma_{ij} [\epsilon_{pq} [u_k^b]] \frac{\partial N^a}{\partial x_j} dV - \int_{S_2} t_i^a N^a \right\} \eta_i^a = 0 + \eta_i^a$$

R_i^a - internal force
on a th node

$$f_i^a$$

System of nonlinear equations for U_k^b

$$R_i^a [U_k^b] = f_i^a$$

Solve with Newton - Raphson

(1) Guess solution $U_k^b = W_k^b$ eg $W_k^b = 0$

(2) Correct $R_i^a [W_k^b + dW_k^b] = f_i^a$

(3) Taylor expansion

$$\left\{ \frac{\partial R_i^a}{\partial U_k^b} dW_k^b = f_i^a - R_i^a [W_k^b] \right.$$

Kaibk

Consistent Tangent or Jacobian

Newton Iteration Loop

$$(1) \text{ Solve } K_{ik} b_k dW_k^b = f_i^a - R_i^a$$

$$(2) W_i^a \rightarrow W_i^a + dW_i^a$$

(3) Check convergence

No

$$\frac{dW_k^b}{W_i^a} \frac{dW_k^b}{W_i^a} \leq tol_1$$

$$\frac{(R_i^a - f_i^a)(R_i^a - f_i^a)}{R_k^b R_k^b} \leq tol_2$$

b Y

(4) Proceed to next time increment

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Newton Loop in EN234 FEA

```
do while (continue_timesteps)
    call assemble_direct_stiffness(fail)

    if (fail) then           ! Force a timestep cutback if stiffness computation fails
        converged=.false.
        iteration = 0
        call compute_static_time_increment(iteration,converged,continue_timesteps, &
            activestateprint,activateuserprint,new_time_increment)
        DTIME = new_time_increment
        dof_increment = 0.d0
        cycle
    endif
    call apply_direct_boundaryconditions
    call solve_direct

    converged = .true.

    do iteration = 1,max_newton_iterations
        call assemble_direct_stiffness(fail)

        converged = .false.
        if (fail) exit

        call apply_direct_boundaryconditions
        call convergencecheck(iteration,converged)
        if (converged) exit
        call solve_direct

    end do

    call compute_static_time_increment(iteration,converged,continue_timesteps, &
        activestateprint,activateuserprint,new_time_increment)

    if (.not.converged) then
        dof_increment = 0.d0
        DTIME = new_time_increment
        cycle
    endif

    if (activestateprint) call print_state
    if (activateuserprint) call user_print(current_step_number)

    !      Update solution and continue
    current_step_number = current_step_number + 1
    dof_total = dof_total + dof_increment
    initial_state_variables = updated_state_variables
    TIME = TIME + DTIME
    DTIME = new_time_increment

end do
```

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Calculating the Jacobian matrix

$$K_{ijkl} = \frac{\partial}{\partial u_k^b} \int_R \sigma_{ij} [\epsilon_{pq}(u_k^b)] \frac{\partial N^a}{\partial x_j} dV$$

$$= \int_R \frac{\partial \sigma_{ij}}{\partial \epsilon_{pq}} \frac{\partial \epsilon_{pq}}{\partial u_k^b} \frac{\partial N^a}{\partial x_j}$$

Recall $\epsilon_{pq} = \frac{1}{2} \left\{ \frac{\partial N^c}{\partial x_q} u_p^c + \frac{\partial N^c}{\partial x_p} u_q^c \right\}$

$$\Rightarrow \frac{\partial \epsilon_{pq}}{\partial u_k^b} = \frac{1}{2} \left\{ \frac{\partial N^b}{\partial x_q} S_{pk} + \frac{\partial N^b}{\partial x_p} S_{qk} \right\}$$

Recall $\frac{\partial \sigma_{ij}}{\partial \epsilon_{pq}} = \frac{\partial \sigma_{ij}}{\partial \epsilon_{qp}}$ \Rightarrow can combine the two terms

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$$\text{Hence } K_{ijkl} = \int_R \frac{\partial \sigma_{ij}}{\partial \varepsilon_{kl}} \frac{\partial N^b}{\partial x_e} \frac{\partial N^q}{\partial x_j} dV$$

Exactly the same as linear elasticity, except σ_{ij} is nonlinear, and C_{ijkl} is now $\partial \sigma_{ij} / \partial \varepsilon_{kl}$

We only need to change calculation for σ and $[D]$

$$[D] = \begin{bmatrix} \frac{\partial \sigma_{11}}{\partial \varepsilon_{11}} & \frac{\partial \sigma_{11}}{\partial \varepsilon_{22}} & \frac{\partial \sigma_{11}}{\partial \varepsilon_{33}} & \frac{\partial \sigma_{11}}{\partial \varepsilon_{12}} & \dots \\ \frac{\partial \sigma_{22}}{\partial \varepsilon_{11}} & \dots & \dots & \dots & \dots \\ \frac{\partial \sigma_{33}}{\partial \varepsilon_{11}} & \dots & \dots & \dots & \dots \end{bmatrix}$$

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In ABAQUS / ENZ34FEA we can use the UMAT subroutine to calculate $\underline{\sigma}$ and $[D]$

ABAQUS / ENZ34FEA will assemble $[k^e]$ handle equation solving, Newton iterations etc

By default elements are B and in ENZ34FEA elements are finite strain

SUBROUTINE UMAT(STRESS,STATEV,DDSDDE,SSE,SPD,SCD,
1 RPL,DDSDDT,DRPLDE,DRPLDT,
2 STRAN,DSTRAN,TIME,DTIME,TEMP,DTEMP,PREDEF,DPRED,CNAME,
3 NDI,NSHR,NTENS,NSTATV,PROPS,NPROPS,COORDS,DROT,PNEWDT,
4 CELENT,DFGRD0,DFGRD1,NOEL,NPT,LAYER,KSPT,KSTEP,KINC)

? Outputs $\text{STRESS}(I) = \underline{\sigma}$

} Input $\text{DDSDDE}(I,J) = [D]$

Other outputs optional

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Computing material tangent stiffness

$$\frac{\partial \sigma_{ij}}{\partial \varepsilon_{ke}}$$

Recall $\sigma_{ij} = S_{ij} + p \delta_{ij}$ $p = K \varepsilon_{qq}$

$$\frac{\partial p}{\partial \varepsilon_{ke}} = K \delta_{qk} \delta_{qe} = K \delta_{ke}$$

$$\begin{aligned} \frac{\partial S_{ij}}{\partial \varepsilon_{ke}} &= \frac{\partial}{\partial \varepsilon_{ke}} \left(\frac{2 \sigma_e (\varepsilon_e)}{2} \frac{\varepsilon_{ij}}{\varepsilon_e} \right) \\ &= \frac{2}{3} \left\{ \frac{\partial \sigma_e}{\partial \varepsilon_e} - \frac{\sigma_e}{\varepsilon_e} \right\} \frac{\partial \varepsilon_e}{\partial \varepsilon_{ke}} \frac{\varepsilon_{ij}}{\varepsilon_e} + \frac{2}{3} \frac{\sigma_e}{\varepsilon_e} \frac{\partial \varepsilon_{ij}}{\partial \varepsilon_{ke}} \end{aligned}$$

$$\varepsilon_e = \sqrt{\frac{2}{3} \varepsilon_{ij} \varepsilon_{ij}} \Rightarrow \frac{\partial \varepsilon_e}{\partial \varepsilon_{pq}} = \frac{2}{3} \frac{\varepsilon_{pq}}{\varepsilon_e}$$

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$$\text{Finally } e_{ij} = \varepsilon_{ij} - \varepsilon_{pp} \delta_{ij}/3$$

$$\frac{\partial e_{ij}}{\partial \varepsilon_{ke}} = \frac{1}{2} (\underbrace{\delta_{ik}\varepsilon_{je} + \delta_{ie}\varepsilon_{jk}}_{} - \frac{1}{3} \delta_{ke} \delta_{ij})$$

Need symmetry in k, l and i, j

$$\frac{d\sigma_{ij}}{d\varepsilon_H} = \frac{4}{9} (E_t - E_s) \frac{e_{ij}e_{kl}}{\varepsilon_e^2} + \frac{2}{3} E_s \left(\frac{\delta_{ik}\delta_{jl} + \delta_{il}\delta_{jk}}{2} - \frac{\delta_{ij}\delta_{kl}}{3} \right) + K \delta_{kl} \delta_{ij}$$

$$E_s = \frac{\sigma_e}{\varepsilon_e} \quad E_t = \frac{d\sigma_e}{d\varepsilon_e}$$

Matrix form: if we define $\underline{e} = [e_{11} \ e_{22} \ e_{33} \ \underline{e}_{12} \ e_{13} \ e_{23}]$

NO 2!!

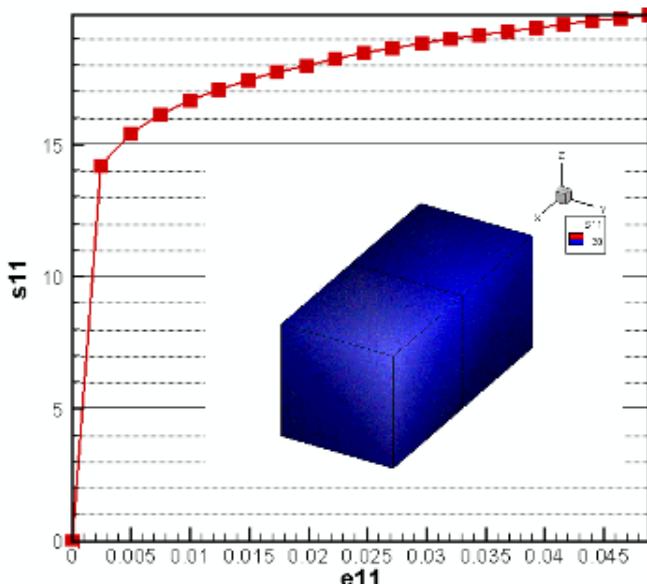
$$[\underline{\underline{\sigma}}] = \frac{4}{9} [E_t - E_s] \underline{e} \otimes \underline{e} + \frac{2}{3} E_s \begin{bmatrix} 2 & 2 & 0 \\ 2 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix} + \left(K - \frac{2}{9} E_s \right) \begin{bmatrix} 1 & 1 & 0 \\ 1 & 1 & 0 \\ 1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

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Example UMAT code

Results

(See code on
Web)



```

s0 = PROPS(1)
e0 = PROPS(2)
n = PROPS(3)
K = PROPS(4)

evol = sum(STRAN(1:3)+DSTRAN(1:3))
edev(1:3) = STRAN(1:3)+DSTRAN(1:3) - evol/3.d0
edev(4:6) = 0.5d0*(STRAN(4:6)+DSTRAN(4:6))

ee = dsqrt(dot_product(edev(1:3),edev(1:3)) +
           2.d0*dot_product(edev(4:6),edev(4:6))/dsqrt(1.5d0))

1   DDSDDE(1:6,1:6) = 0.d0
if (ee<=e0) then
  se = s0*( dsqrt( (1.d0+n*n)/((n-1.d0)*(n-1.d0)) -
    (n/(n-1.d0) - ee/e0)**2.d0 ) - 1.d0/(n-1.d0) )
  dsdedee = (s0/e0)*(n/(n-1.d0) - ee/e0)/
1   dsqrt((1.d0+n*n)/((n-1.d0)*(n-1.d0)) - (n/(n-1.d0)-ee/e0)**2.d0)
  if (ee==0.d0) then
    stress = 0.d0
    Es = n*s0/e0
  else
    Es = se/ee
    Et = dsdedee
    stress = 2.d0*se*edev/(3.d0*ee)
    stress(1:3) = stress(1:3) + K*evol
    DDSDDE(1:6,1:6) =
      4.d0*(Et-Es)*spread(edev,dim=2,ncopies=6)*
                  spread(edev,dim=1,ncopies=6)/(9.d0*ee*ee)
  endif
else
  se = s0*(ee/e0)**(1.d0/n)
  stress = 2.d0*se*edev/(3.d0*ee)
  stress(1:3) = stress(1:3) + K*evol
  Et = se/(n*ee)
  Es = se/ee
  DDSDDE(1:6,1:6) = 4.d0*(Et-Es)*spread(edev,dim=2,ncopies=6)*
                  spread(edev,dim=1,ncopies=6)/(9.d0*ee*ee)
endif
forall(j=1:3) DDSDDE(j,j) = DDSDDE(j,j) + 2.d0*Es/3.d0
forall(j=4:6) DDSDDE(j,j) = DDSDDE(j,j) + Es/3.d0
DDSDDE(1:3,1:3) = DDSDDE(1:3,1:3) + (K-2.d0*Es/9.d0)

```