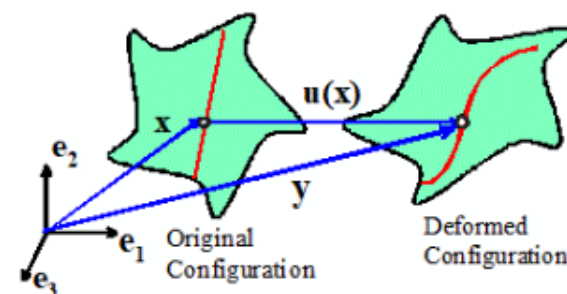


Review – FEA for finite deformation - hyperelasticity

Kinematics

$$y_i = x_i + u_i(x_k) \quad F_{ij} = \delta_{ij} + \frac{\partial u_i}{\partial x_j}$$

$$J = \det(\mathbf{F}), \quad B_{ij} = F_{ik} F_{jk}$$



Equilibrium

$$\int_R \sigma_{ji} \frac{\partial \eta_i}{\partial y_j} dV_0 - \int_{S_2} t_i \eta_i dA = 0 \quad \forall \text{ admiss } \eta_i \Leftrightarrow \partial \sigma_{ij} / \partial y_i = 0 \quad n_i \sigma_{ij} = t_j^* \text{ on } S_2$$

Shear Modulus

Bulk Modulus

Stress-strain relation

$$\sigma_{ij} = \frac{\mu_1}{J^{5/3}} \left(B_{ij} - \frac{1}{3} B_{kk} \delta_{ij} \right) + K_1 (J - 1) \delta_{ij} \quad \tau_{ij} = J \sigma_{ij}$$

Discrete equilibrium eq

$$R_i^a [u_k^b] = f_i^a \quad R_i^a = \int_{V_0} \frac{\partial N^a}{\partial y_j} \tau_{ji} dV_0 \quad f_i^a = \int_{S_2^0} t_i^{0*} N^a dA_0$$

Solve nonlinear equilibrium eq with Newton-Raphson

Newton correction equation

$$K_{aibk} dw_k^b = f_i^a - R_i^a [w_k^b]$$

$$K_{aibk} = \int_{V_0} \frac{\partial N^a}{\partial y_j} \frac{\partial \tau_{ji}}{\partial B_{pq}} (\delta_{pk} B_{ql} + \delta_{qlk} B_{pl}) \frac{\partial N^b}{\partial y_l} dV_0 - \int_{V_0} \frac{\partial N^a}{\partial y_k} \tau_{ji} \frac{\partial N^b}{\partial y_j} dV_0$$

$$\frac{\partial \tau_{ij}}{\partial B_{kl}} = \frac{\mu_1}{J^{2/3}} \left(\frac{\delta_{ik} \delta_{jl} + \delta_{jk} \delta_{il}}{2} - \frac{1}{3} \delta_{kl} \delta_{ij} \right) - \frac{1}{3} \frac{\mu_1}{J^{2/3}} \left(B_{ij} - \frac{1}{3} B_{mm} \delta_{ij} \right) B_{kl}^{-1} + K_1 (J^2 - \frac{1}{2} J) \delta_{ij} B_{kl}^{-1}$$

FE implementation

$$[K^{el}] = \int_{\Omega_{eo}} [B]^T [D] [G] [B^*] dV_0 - \int_{\Omega_{eo}} [\gamma] dV_0$$

$[B]$ - usual $[B]$ (6×3 NNodes) - except $\partial N / \partial x$ now $\partial N / \partial y$

$$[D] = \begin{bmatrix} \partial \tau_{11} / \partial B_{11} & \partial \tau_{11} / \partial B_{22} & \partial \tau_{33} / \partial B_{33} \\ \partial \tau_{22} / \partial B_{11} & & \\ \vdots & & \end{bmatrix} \quad \bullet \text{ Not symmetric}$$

Define $\underline{b} = [B_{11} \ B_{22} \ B_{33} \ B_{12} \ \dots]$

$$\underline{b}^{-1} = [B_{11}^{-1} \ B_{22}^{-1} \ \dots]$$

$$\underline{I} = [1, 1, 1, 0, 0, 0]$$

page 3 Then:

$$\mathbf{D} = \frac{\mu_1}{J^{2/3}} \begin{bmatrix} 1 & & & & 0 \\ & 1 & & & \\ & & 1 & & \\ & & & 1/2 & \\ 0 & & & & 1/2 \end{bmatrix} + \frac{\mu_1}{3J^{2/3}} \left(\frac{B_{nn}}{3} \underline{I} \otimes \underline{b}^{-1} - \underline{I} \otimes \underline{I} - \underline{b} \otimes \underline{b}^{-1} \right) + K_1 J (J - 1/2) \underline{I} \otimes \underline{b}^{-1}$$

$[\mathbf{B}^*]$ is a modified $[\mathbf{B}]$ that maps $\underline{u}$ onto displacement grads

$$\underline{g} = [\mathbf{B}^*] \underline{u}^{el}$$

$$\underline{g} = \left[\frac{\partial u_1}{\partial y_1} \quad \frac{\partial u_2}{\partial y_2} \quad \frac{\partial u_3}{\partial y_3} \quad \frac{\partial u_1}{\partial y_2} \quad \frac{\partial u_2}{\partial y_1} \quad \frac{\partial u_1}{\partial y_3} \quad \frac{\partial u_3}{\partial y_1} \quad \frac{\partial u_2}{\partial y_3} \quad \frac{\partial u_3}{\partial y_2} \right]^T$$

page 3

Then $[B^*]$ has form

$$G = \begin{bmatrix} 2B_{11} & 0 & 0 & 2B_{12} & 0 & 2B_{13} & 0 & 0 & 0 \\ 0 & 2B_{22} & 0 & 0 & 2B_{12} & 0 & 0 & 2B_{23} & 0 \\ 0 & 0 & 2B_{33} & 0 & 0 & 0 & 2B_{13} & 0 & 2B_{13} \\ 2B_{12} & 2B_{12} & 0 & 2B_{22} & 2B_{11} & 2B_{23} & 0 & 2B_{13} & 0 \\ 2B_{13} & 0 & 2B_{13} & 2B_{23} & 0 & 2B_{33} & 2B_{11} & 0 & 2B_{12} \\ 0 & 2B_{23} & 2B_{23} & 0 & 2B_{13} & 0 & 2B_{12} & 2B_{33} & 2B_{22} \end{bmatrix}$$

$$B^* = \begin{bmatrix} \frac{\partial N^1}{\partial y_1} & 0 & 0 & \frac{\partial N^2}{\partial y_1} & 0 & 0 \\ 0 & \frac{\partial N^1}{\partial y_2} & 0 & 0 & \frac{\partial N^2}{\partial y_2} & 0 \\ 0 & 0 & \frac{\partial N^1}{\partial y_3} & 0 & 0 & \frac{\partial N^2}{\partial y_3} \\ \frac{\partial N^1}{\partial y_2} & 0 & 0 & \frac{\partial N^2}{\partial y_2} & 0 & 0 \\ 0 & \frac{\partial N^1}{\partial y_1} & 0 & 0 & \frac{\partial N^2}{\partial y_1} & 0 \dots \\ \frac{\partial N^1}{\partial y_3} & 0 & 0 & \frac{\partial N^2}{\partial y_3} & & \\ 0 & 0 & \frac{\partial N^1}{\partial y_1} & 0 & 0 & \frac{\partial N^2}{\partial y_1} \\ 0 & \frac{\partial N^1}{\partial y_3} & 0 & & & \\ 0 & 0 & \frac{\partial N^1}{\partial y_2} & & & \end{bmatrix}$$

Geometric Stiffness

$$\gamma_{aibk} = \frac{\partial N^a}{\partial y_k} r_{ji} \frac{\partial N^b}{\partial y_j}$$

Define $P_{ak} = \frac{\partial N^a}{\partial y_k}$

$n \times 3$

$S_{bi} = \tau_{ji} \frac{\partial N^b}{\partial y_j}$

$n \times 3$

Then

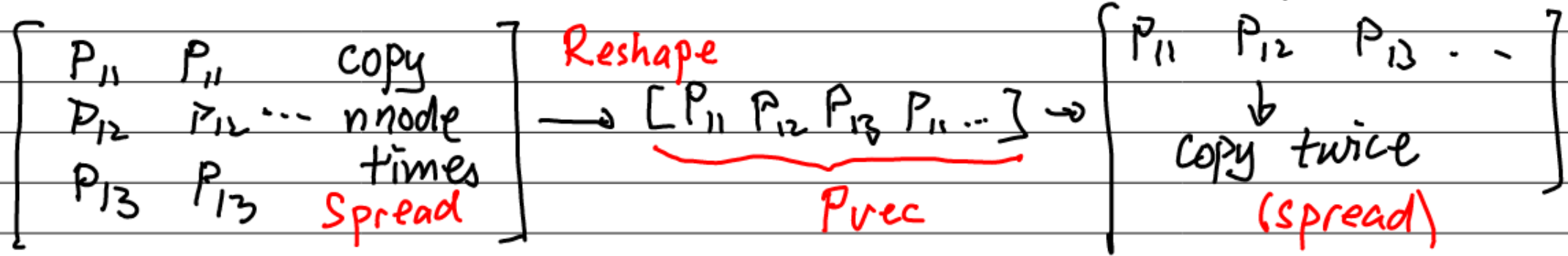
$$[Y] = \begin{bmatrix} P_{11} S_{11} & P_{12} S_{11} & P_{13} S_{11} & P_{11} S_{21} & \dots \\ P_{11} S_{12} & & & & \\ P_{11} S_{13} & & & & \\ P_{21} S_{11} & & & & \end{bmatrix}$$

This is

$$[Y] = \begin{bmatrix} P_{11} & P_{12} & P_{13} & \dots \\ P_{11} & P_{12} & P_{13} & \dots \\ P_{11} & P_{12} & P_{13} & \dots \\ P_{21} & P_{22} & P_{23} & \dots \end{bmatrix} \odot \begin{bmatrix} S_{11} & S_{11} & S_{11} & S_{21} & \dots \\ S_{12} & S_{12} & S_{12} & & \\ S_{13} & S_{13} & S_{13} & & \end{bmatrix}$$

elemental product

We can assemble $[P]$, $[S]$ with 'spread', reshapes



Geometric stiffness

```
S = reshape(matmul(transpose(Bbar), stress), (/3, 3*NNODE/3/))
```

```
do i = 1, NNODE
```

1

```
Pvec(1:3*NNODE) = reshape(spread(transpose(dNdy(i:i, 1:3)),  
                                dim=2, ncopies=NNODE), (/3*NNODE/))
```

← Pvec

```
Pmat(3*i-2:3*i, 1:3*NNODE) = spread(Pvec, dim=1, ncopies=3)
```

← Copy to create 3 rows of $[P]$

1

```
Svec(1:3*NNODE) = reshape(spread(S(1:3, i:i),  
                                dim=2, ncopies=NNODE), (/3*NNODE/))
```

```
Smat(3*i-2:3*i, 1:3*NNODE) = spread(Svec, dim=1, ncopies=3)
```

```
end do
```

```
AMATRIX(1:3*NNODE, 1:3*NNODE) = AMATRIX(1:3*NNODE, 1:3*NNODE) -
```

1

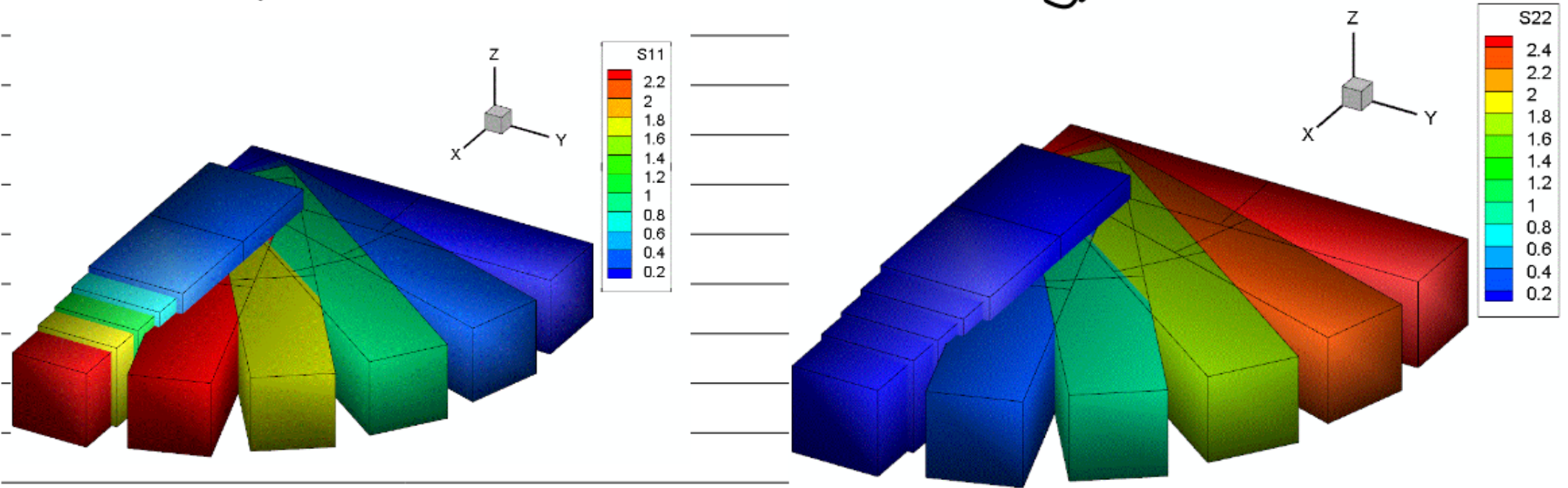
```
Pmat(1:3*NNODE, 1:3*NNODE) * transpose(Smat(1:3*NNODE, 1:3*NNODE))
```

2

```
*w(kint)*determinant
```

← Elemental product

Sample Results - Stretch / Rotate a hyperelastic bar



Remarks: For actual applications elastomers are near incompressible $\Rightarrow$ B-bar or hybrid elements
See HW #6 2015 for B-bar implementation

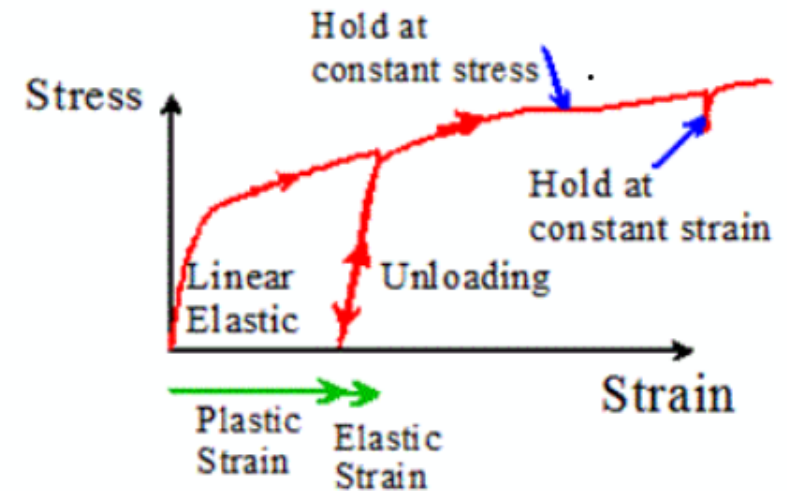
In ABAQUS best not to use UMAT for finite strain elasticity - use UHYPER instead

7.3) FEA for nonlinear & history dependent materials - small strain viscoplasticity

Assume infinitesimal deformations

Constitutive law

$$\dot{\epsilon}_{ij} = \dot{\epsilon}_{ij}^e + \dot{\epsilon}_{ij}^p$$



Elastic strains

$$\dot{\epsilon}_{ij}^e = \frac{1+\nu}{E} \dot{S}_{ij} + \frac{1-2\nu}{E} \dot{\sigma}_{kk} \frac{\delta_{ij}}{3}$$

$$S_{ij} = \sigma_{ij} - \sigma_{kk} \delta_{ij} / 3$$

Plastic strain rate:

- (1) Isotropic material
- (2) Plastic strains volume preserving
- (3) Assume plastic strain rate parallel to deviatoric stress "associated flow"

$$\text{Let } \sigma_e = \sqrt{\frac{3}{2} S_{ij} S_{ij}} \quad \text{Von Mises stress}$$

$$\dot{\epsilon}_{ij}^p = \dot{\epsilon}_0 \left(\frac{\sigma_e}{\sigma_0} \right)^m \frac{3}{2} \frac{S_{ij}}{\sigma_e} \quad \dot{\epsilon}_0, m \text{ material props}$$

$$\sigma_0 = Y \left(1 + \frac{\epsilon_e}{\epsilon_0} \right)^{1/n} \quad Y, n, \epsilon_0$$

ϵ_e = accumulated plastic strain magnitude

$$\epsilon_e = 0 \text{ @ } t = 0$$

$$\dot{\epsilon}_e = \sqrt{\frac{2}{3} \dot{\epsilon}_{ij}^p \dot{\epsilon}_{ij}^p}$$

FEA implementation :

- (1) Time dependent material
- (2) History dependent - need to track load history

Address these with simple time-stepping

(1) Assume $U_i = 0$ $\epsilon_e = 0$ @ $t = 0$

(2) Consider time increment Δt

(3) Find ΔU_i , $\Delta \epsilon$ by satisfying equilibrium at end of step

$$U_i \rightarrow U_i + \Delta U_i \quad \epsilon_e \rightarrow \epsilon_e + \Delta \epsilon_e$$

Repeat.

Equilibrium equation (weak form) has usual form

$$\int_{\mathcal{R}} \sigma_{ij} [\Delta \varepsilon_{pq}(\Delta u_k), \Delta \varepsilon_e, \Delta t] \frac{\partial \eta_i}{\partial x_j} dV - \int_{S_2} t_i^* \eta_i = 0 \quad \forall \text{admiss } \eta_i$$

$$u_i + \Delta u_i = u_i^*(t + \Delta t) \quad \text{on } S_1$$