

School of Engineering Brown University **EN40: Dynamics and Vibrations** 

Final Examination Fri May 10 2019: 9am-12:00pm

NAME:

## **General Instructions**

- No collaboration of any kind is permitted on this examination.
- You may bring 2 double sided pages of reference notes. No other material may be consulted
- Write all your solutions in the space provided. No sheets should be added to the exam.
- Make diagrams and sketches as clear as possible, and show all your derivations clearly. Incomplete solutions will receive only partial credit, even if the answer is correct.
- If you find you are unable to complete part of a question, proceed to the next part.

## Please initial the statement below to show that you have read it

'By affixing my name to this paper, I affirm that I have executed the examination in accordance with the Academic Honor Code of Brown University. **PLEASE WRITE YOUR NAME ABOVE ALSO!** 

1-20 [40 points]	
21 [10 POINTS]	
22 [10 POINTS]	
TOTAL [60 POINTS]	

# FOR PROBLEMS 1-20 WRITE YOUR ANSWER IN THE SPACE PROVIDED. ONLY THE ANSWER APPEARING IN THE SPACE PROVIDED WILL BE GRADED. ILLEGIBLE ANSWERS WILL NOT RECEIVE CREDIT.

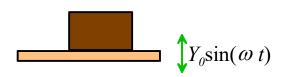
**1.** The platform shown in the figure vibrates in the vertical direction with a harmonic displacement  $y(t) = Y_0 \sin \omega t$ . The amplitude of its vertical acceleration is

- (a)  $Y_0 \omega^2$
- (b)  $Y_0 / \omega^2$
- (c)  $Y_0 \omega$
- (d)  $Y_0 / \omega$
- (e) None of the above

2. The platform shown in the figure vibrates in the vertical direction with a harmonic displacement  $y(t) = Y_0 \sin \omega t$ , with  $\omega = 100$  rad/s. For each value of the amplitude listed below, indicate (by circling the appropriate response) whether the mass will remain in contact with the platform





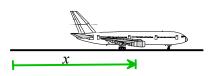


(a) 0.01mm	Remains in contact	Loses contact	
(b) 0. 1mm	Remains in contact	Loses contact	
(c) 1mm	Remains in contact	Loses contact	
(d) 10mm	Remains in contact	Loses contact	

(2 POINTS)

**3.** The aircraft shown in the figure starts at rest and has and acceleration that depends on its speed v as  $a = A_0 \{1 - (v^2 / v_0^2)\}$ , where  $A_0$  and  $v_0$  are constants. After the aircraft has traveled a distance *x*, its speed is

(a) 
$$v = \sqrt{2A_0x}$$
  
(b)  $v = \sqrt{2A_0x\{1 - (v^2 / v_0^2)\}}$   
(c)  $v = v_0\sqrt{1 - \exp(-2A_0x / v_0^2)}$   
(d)  $v = v_0\sqrt{1 - \exp(-A_0x / v_0^2)}$   
(e) None of the above



ANSWER (2 POINTS)

**4.** The trajectory of a particle is specified in polar coordinates as a function of time as

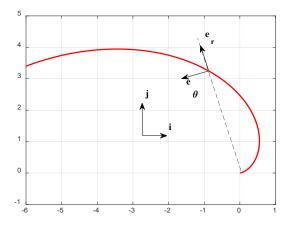
$$r = t^2$$
  $\theta = t$ 

At time t=1s, the particle's velocity vector is

(a) 
$$\mathbf{v} = \mathbf{e}_r + \mathbf{e}_{\theta}$$

(b) 
$$\mathbf{v} = \frac{1}{2}\mathbf{e}_r + \mathbf{e}_{\theta}$$

- (c) v = 2i + j
- (d)  $\mathbf{v} = 2\mathbf{e}_r + \mathbf{e}_{\theta}$
- (e) None of the above



5. The trajectory of a particle is specified in polar coordinates as a function of time as

 $r = t^2$  $\theta = t$ 

At time t=1s, the particle's acceleration vector is

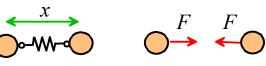
- (a)  $\mathbf{a} = 2\mathbf{e}_r + 4\mathbf{e}_{\theta}$
- (b)  $\mathbf{a} = \mathbf{e}_r + 2\mathbf{e}_{\theta}$ (c)  $\mathbf{a} = \mathbf{e}_r + 4\mathbf{e}_{\theta}$
- (d)  $\mathbf{a} = -\mathbf{e}_r$
- (e) None of the above

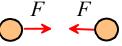
ANSWER (2 POINTS)

6. Two dashpots connected in series are stretched by a constant force F as shown in the figure. The rate of work done by the force is (a) *cF* (b)  $2F^2 / c$ (c) F/c(d)  $F^2/(2c)$ (e) None of the above ANSWER (2 POINTS)

7. The bond between atoms in a diatomic molecule has a potential energy given by

$$U = -F_0 d \left\{ \left(\frac{x}{d}\right)^{-2} - \left(\frac{x}{d}\right)^{-4} \right\}$$





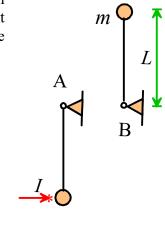
where x is the separation between the atoms and  $F_0$ and *d* are constants. When x = 2d the attractive force between the atoms is

(a) 
$$F = F_0 / 2$$
  
(b)  $F = F_0 / 4$   
(c)  $F = F_0 / 8$   
(d)  $F = F_0 / 16$   
(e) None of the above

ANSWER\_\_\_\_\_(2 POINTS)

8. At time t=0 the pendulum shown in the figure is at rest. It is then subjected to a horizontal impulse with magnitude *I*. During its subsequent motion, it just comes to rest in the inverted configuration shown. The magnitude of the impulse is

(a)  $I = m\sqrt{2gL}$ (b) I = 2mgL(c) I = mgL(d)  $I = 2m\sqrt{gL}$ (e) None of the above

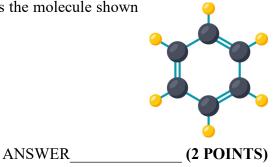


(2 POINTS) ANSWER

9. How many degrees of freedom and vibration modes has the molecule shown in the figure?

(a) 18 DOF and 24 vibration modes (b) 24 DOF and 18 vibration modes (c) 36 DOF and 30 vibration modes (d) 30 DOF and 36 vibration modes

- (e) None of the above

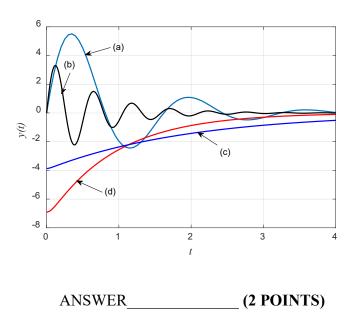


**10.** Which curve in the figure shows a solution to the differential equation

$$\frac{d^2y}{dt^2} + 2\frac{dy}{dt} + 16y = 0$$

Select (a)-(d) or

(e) none of the curves are a correct solution



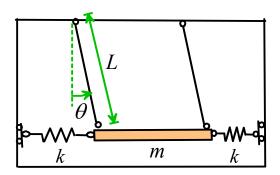
**11.** The spring-pendulum vibration isolation system shown in the figure has an equation of motion

$$mL^2 \frac{d^2\theta}{dt^2} + 2kL^2 \sin\theta \cos\theta + mgL\sin\theta = 0$$

The natural frequency of small amplitude oscillations is

(a) 
$$\sqrt{\frac{2k}{m} + \frac{g}{L}}$$
  
(b)  $\sqrt{\frac{2k}{m}}$   
(c)  $\sqrt{\frac{2k}{m} + \frac{2g}{L}}$   
(d)  $\sqrt{\frac{g}{L}}$ 

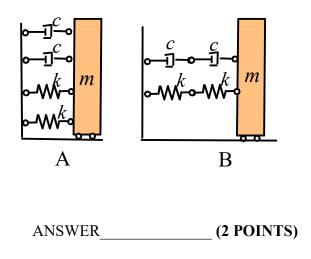
(e) None of the above



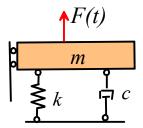
ANSWER\_\_\_\_\_(2 POINTS)

12. System A has natural frequency and damping coefficient  $\omega_n, \zeta$ . System B has natural frequency and damping coefficient

- (a)  $\omega_{nB} = \omega_n / \sqrt{2}$ ,  $\zeta_B = \zeta / \sqrt{2}$ (b)  $\omega_{nB} = \omega_n / 2$ ,  $\zeta_B = \zeta / 2$ (c)  $\omega_{nB} = 2\omega_n$ ,  $\zeta_B = 2\zeta$
- (d)  $\omega_{nB} = \omega_n / 2$ ,  $\zeta_B = \zeta / \sqrt{2}$
- (e) None of the above



13. The spring-mass system shown in the figure has undamped natural frequency  $\omega_n$  and damping coefficient  $\zeta$ . It is subjected to a harmonic force with amplitude  $F_0$  and frequency equal to its undamped natural frequency ( $\omega = \omega_n$ ). The amplitude of vibration of the mass is

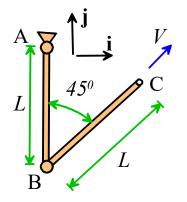


(a) 
$$Y_0 = \frac{F_0}{2\zeta}$$
  
(b) 
$$Y_0 = \frac{F_0}{c} \sqrt{\frac{m}{k}}$$
  
(c) 
$$Y_0 = \frac{F_0}{c} \sqrt{\frac{k}{m}}$$
  
(d) 
$$Y_0 = \frac{F_0}{k}$$

8

14. Point A on the actuator shown in the figure is stationary, and point C moves at constant speed V along a straight line parallel to the vector  $\mathbf{i} + \mathbf{j}$ . The angular velocities of arms AB and BC are

- (a)  $\omega_{BC} = V / (L\sqrt{2})$   $\omega_{AB} = V / L$ (b)  $\omega_{BC} = V / L \quad \omega_{AB} = -V / (L\sqrt{2})$ (c)  $\omega_{BC} = V / L$   $\omega_{AB} = \sqrt{2}V / L$
- (d)  $\omega_{BC} = \sqrt{2}V/L$   $\omega_{AB} = V/L$
- (e) None of the above

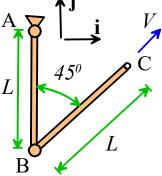


ANSWER (2 POINTS)

15. Point A on the actuator shown in the figure is stationary, and point C moves at constant speed V along a straight line parallel to the vector  $\mathbf{i} + \mathbf{j}$ . The angular accelerations of arms AB and BC are

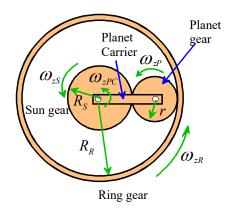
- (a)  $\alpha_{BC} = (1 \sqrt{2})V^2 / L^2$   $\alpha_{AB} = (\sqrt{2} 1)V^2 / L^2$ (b)  $\alpha_{BC} = V^2 / L^2$   $\alpha_{AB} = \sqrt{2}V^2 / L^2$ (c)  $\alpha_{BC} = (1 - 2\sqrt{2})V^2 / L^2$   $\alpha_{AB} = (\sqrt{2} - 2)V^2 / L^2$ (d)  $\alpha_{BC} = -2\sqrt{2}V^2 / L^2 \qquad \alpha_{AB} = V^2 / (\sqrt{2}L^2)$
- (e) None of the above

L В



16. In the figure shown, the planet carrier (the bar) rotates counterclockwise with angular speed  $\omega_{zPC} = \omega_0$ . The planet gear has zero angular speed. The angular speed of the sun gear is

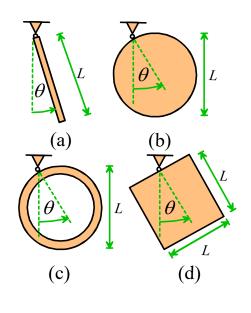
- (a)  $\omega_0(R_s r) / R_s$  clockwise
- (b)  $\omega_0(R_s r)/R_s$  counterclockwise
- (c)  $\omega_0(R_s+r)/R_s$  clockwise
- (d)  $\omega_0(R_s + r) / R_s$  counterclockwise
- (e) None of the above



ANSWER (2 POINTS)

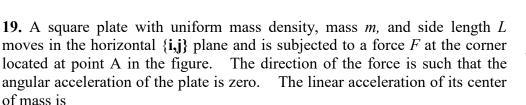
**17.** The four objects shown in the figure are suspended from frictionless pivots, and set swinging (as pendula) with the same (small) amplitude. Which pendulum has the highest natural frequency?

- (a) The rod
- (b) The disk
- (c) The ring
- (d) The square
- (e) All four have the same frequency.



18 A two bladed wind-turbine with total mass 120kg (60kg per blade) and rotor diameter 20m is spun up from rest to an angular speed of 10 radians per second in 100 sec. The (constant) torque exerted by the wind on the turbine is

(a) 200 Nm (b) 400 Nm (c) 800 Nm (d) 1000 Nm (e) None of the above



(a)  $(F/m)(i+j)/\sqrt{2}$ (b)  $(2F/(3m)(i+j)/\sqrt{2})$ 

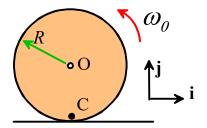
(c) 
$$(F/(4m)(i+j)/\sqrt{2})$$

(d) (*F* / *m*)**i** 

(e) None of the above

ANSWER (2 POINTS)

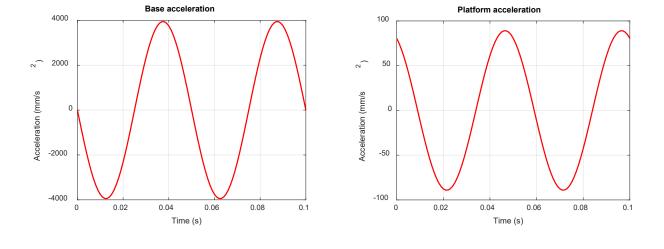
**20.** At time t=0 the cylinder shown in the figure has angular velocity  $\mathbf{\omega} = \omega_0 \mathbf{k}$  with  $\omega_0 > 0$  and its center is stationary. It is in contact with a stationary surface. The contact between the cylinder and surface has friction coefficient  $\mu$ . Identify whether the statements below are true or false



# AT TIME *t*=θ:

(a) No slip occurs at the contact	Т	F
(b) The friction force on the cylinder has magnitude $\mu mg$	Т	F
(c) The friction force on the cylinder acts to the right	Т	F
(d) The cylinder has acceleration $\mathbf{a}_G = -\mu g \mathbf{i}$	Т	F

(2 POINTS)



**21** The figure shows readings from accelerometers attached to the base and platform of a vibration isolation table.

21.1 Find

(a) The period of the vibration;

Platform

С

Base

т

k

(b) The frequency of the vibration (in both Hz and radians per second);

[1 POINT]

(c) The amplitude of the displacement (in mm) of the base

# [1 POINT]

(d) The amplitude of the displacement (in mm) of the platform

(e) The magnification.

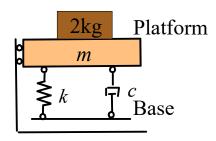
# [1 POINT]

[1 POINT]

21.2 In a separate experiment, the natural frequency of vibration of the table is measured to be 2Hz. Use this information and your solution to 21.1 to calculate the damping coefficient  $\zeta$  for the platform

[3 POINTS]

21.3 A 2kg mass is placed on the table. It is found that the natural frequency of the table with the mass on its surface decreases to 1Hz. Calculate the value of the table mass m, the spring constant k and the dashpot coefficient c.

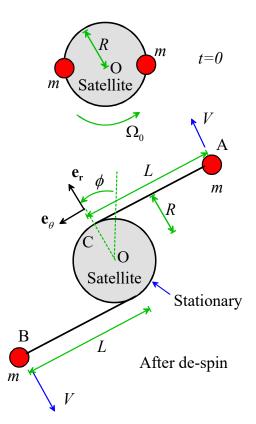


[2 POINTS]

**22** The figure shows a so called 'yoyo de-spin' of a satellite. The satellite is cylindrical, with radius R and mass moment of inertia about its center  $I_{Gzz}$ . Two masses m (with negligible mass moment of inertia about their COM) are attached to the cylinder by inextensible, massless tethers.

At time t=0 the tethers are wound tightly around the cylinder and the assembly spins with angular speed  $\Omega_0$ . To de-spin the satellite, the masses are released, and the rotational motion of the assembly causes the tethers to unwind from the cylinder. This slows the rotation of the cylinder. When the rotation of the cylinder stops, the tethers are cut. The goal of this problem is to determine the length of the tethers necessary to stop the rotation of the satellite.

22.1 Consider the system at time t=0. Assume that the center of the satellite at O is stationary and the satellite (with masses attached) spins at angular speed  $\Omega_0$ . Find the total kinetic energy of the system (small masses + cylinder together) at time t=0, in terms of  $\Omega_0 m, R, I_{Gzz}$ . Treat the small masses m as particles.



#### [2 POINTS]

22.2 Find a formula the total angular momentum of the system (masses + cylinder) about the center of the cylinder at time t=0 in terms of  $\Omega_0 m, R, I_{Gzz}$ .

## [1 POINT]

22.3 Consider the assembly at the instant that the cylinder just comes to rest (after de-spin). Write down the position vector of the mass at A at this instant, (taking the origin to be at O) in the  $\{\mathbf{e}_r, \mathbf{e}_{\theta}\}$  basis in terms of *R*, *L*.

# [1 POINT]

22.4 Suppose that at the instant the satellite stops rotating, the tether CA has angular speed  $\omega = d\phi/dt$ and length *L*. Note that the cylinder (i.e. the satellite) and tether must have the same velocity where they touch at point C. Use the rigid body kinematics formula to find the velocity vector of the mass at A in the  $\{\mathbf{e}_r, \mathbf{e}_{\theta}\}$  basis in terms of  $\omega, L$ .

#### [1 POINT]

22.5 Hence, find formulas for the total kinetic energy and angular momentum about O of the system (i.e. the satellite and both masses combined) at the instant that the cylindrical satellite comes to rest, in terms of  $\omega$ , *L*, *m*.

# [2 POINTS]

22.6 Finally, by considering the energy and angular momentum of the system show that the cylinder comes to rest when the tether length reaches

$$L = \sqrt{\frac{\left(I_{Gzz} + 2mR^2\right)}{2m}}$$

[3 POINTS]